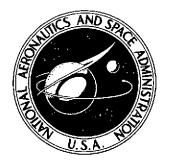
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AN INTEGRAL EQUATION FORMULATION FOR PREDICTING RADIATION PATTERNS OF A SPACE SHUTTLE ANNULAR SLOT ANTENNA

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16. Abstract

An integral equation formulation is applied to predict pitch- and roll-plane radiation patterns of a thin VHF/UHF (very high frequency/ultra high frequency) annular slot communications antenna operating at several locations in the nose region of the Space Shuttle orbiter. Digital computer programs used to compute radiation patterns are given and the use of the programs is illustrated. Experimental verification of computed patterns is given from measurements made on 1/35-scale models of the orbiter.

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AN INTEGRAL EQUATION FORMULATION FOR PREDICTING RADIATION PATTERNS OF A SPACE SHUTTLE ANNULAR SLOT ANTENNA¹

By J. Earl Jones and J. H. Richmond² Langley Research Center

SUMMARY

A synopsis is presented of a recently developed, integral equation formulation for predicting radiation patterns of thin axial slot antennas mounted on an infinitely long, perfectly conducting cylinder of arbitrary cross section. The method is applied to compute radiation patterns in both the pitch and roll planes of a thin VHF/UHF (very high frequency/ultra high frequency) annular slot communications antenna operating at several locations in the nose region of the Space Shuttle orbiter. Digital computer programs used to compute radiation patterns are also given and the use of the programs is illustrated. Experimental verification of computed patterns is given from measurements made on 1/35-scale models of the orbiter.

The results indicate that even when formulated for two-dimensional bodies, the integral equation formulation is useful for predicting principal plane radiation patterns of antennas mounted on moderate to large electrical size bodies such as the Space Shuttle orbiter. The upper limit of the perimeter of the body is dependent on the size of the digital computer used to make the computations. For the computer employed, the maximum perimeter was found to be from 30 to 36 wavelengths.

INTRODUCTION

The Space Shuttle is a reusable NASA vehicle that is expected to receive considerable attention in the forthcoming decade. The motivation for the development of the Space

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Shuttle is the reduction in cost of launching spacecraft into earth orbit and the transportation of men and supplies to and from orbiting space stations. Reusability is achieved by allowing the vehicle to take off like a rocket and to land like an airplane. Thus, the Space Shuttle will require a large number of antennas, since aircraft-type landing and navigation antennas will be needed in addition to the usual spacecraft antennas. Moreover, since the vehicle surface will be at extreme temperatures during atmospheric reentry, antenna locations must be selected in regions where minimum temperatures are anticipated, as well as in regions where structural impact is minimal.

Space Shuttle antennas must possess specific performance characteristics, that is, radiation patterns, impedance, and so forth, in order for desired mission objectives to be achieved. A purely experimental design approach for one antenna would be to (a) construct many scale models of the orbiter, each model containing a proposed antenna geometry at a proposed location, (b) conduct performance measurements, and (c) conclude by selecting the geometry and location giving the most desirable data. However, for Space Shuttle antenna design, this approach has two disadvantages: (a) excessive time and cost are required to construct many models and to conduct many measurements, and (b) changes in the geometry of the vehicle, which is currently only in early design stages, would necessitate the construction and testing of new models to determine whether the antenna performance remains adequate.

A more desirable approach to Space Shuttle antenna design would be to apply theoretical methods which would provide some indication of antenna performance. Easily adaptable to a digital computer, such methods could then be used to compute antenna performance for a wide variety of conditions of antenna geometry and location on the vehicle surface. The design parameters yielding desirable theoretical performance data could then be used to construct a scale model for final checkout. Not only are excessive model construction and testing avoided in this approach, but also the time required to ascertain the influence of a vehicle geometry design change is significantly reduced.

The two basic theoretical methods believed to provide the most useful insight into Space Shuttle antenna design are (a) the integral equation formulation (IEF) (refs. 1 to 5) and (b) the geometrical theory of diffraction (GTD) (refs. 6 to 9). Both IEF (refs. 10 to 12) and GTD (refs. 13 to 16) have recently been used with success in the design of spacecraft and aircraft antennas, and the advantages and limitations of each method are well-documented (ref. 17). Even though GTD and IEF are inherently useful for analyzing antennas mounted on electrically large and electrically small bodies, respectively, the nature of the Space Shuttle antenna design will require the application of both methods. Furthermore, since there is usually a range of body electrical size for which both GTD and IEF are applicable (ref. 17), the use of both methods serves as a check case.

In this paper, an IEF recently developed by Richmond (ref. 5) hereafter designated as the "Richmond Integral Equation Formulation" (RIEF), is briefly described and applied to predict radiation patterns in both the pitch and roll planes of a thin VHF/UHF annular slot communications antenna operating at several locations in the nose region of the Space Shuttle orbiter. Radiation patterns computed by RIEF are given for selected cases. Experimental verification for RIEF is given from radiation pattern measurements made with 1/35-scale models of the Space Shuttle orbiter, and the merits and limitations of RIEF for Space Shuttle antenna design are discussed.

The digital computer programs used for computing radiation patterns are given in appendix A. Finally, an example illustrating the use of the programs is given in appendix B.

SYMBOLS

| a | mean radius of annular slot |
|---------------------------------------|--|
| ₫Į | inner diameter of annular slot |
| $\mathtt{d}_{\boldsymbol{M}}$ | mean diameter of annular slot |
| d_n | length of segment n |
| ^d O | outer diameter of annular slot |
| Ē | electric field vector in annular slot aperture |
| $\overline{\mathbf{E}}_{\mathbf{i}}$ | electric field radiated by strip vee dipole i |
| \mathtt{E}_{ϕ} | radiation field |
| [1] | $N \times 1$ (column) matrix of unknown dipole mode currents |
| In | current per unit length at point Pn |
| i | index of point, segment, strip vee dipole |
| $\overline{\mathrm{J}}_{\mathrm{Dn}}$ | surface current density along strip vee dipole n |

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surface current density along segment n
\widetilde{\mathbf{J}}_{\mathbf{n}}
             index of point, segment, strip vee dipole
j
             free-space propagation constant
k
             number of points, number of segments, number of strip vee dipoles,
Ν
               number of simultaneous equations
             number of coarse points
N١
             index of point, segment, strip vee dipole
n
                    dimensioning integers
n_1, n_2, n_3, n_4, n_5
             point of intersection between segments n-1 (segment N for n=1)
\mathbf{P}_{\mathsf{n}}
               and n
ŝ'n
             orientation unit vector of segment n
             distance from point P_{j-1} along strip vee dipole j
t_i
             distance along segment n from point Pn
t_n
             distance along segment n-1 from point P_{n-1}
t_{n-1}
[V]
             N \times 1 (column) excitation matrix
V_n
             voltage of point P_n
             distances along X- and Y-axes
x, y
\hat{x}, \hat{y}
             vector notation for X- and Y-axes
             distances of point P_n along X- and Y-axes
x_n, y_n
[Z]
             N \times N matrix of impedance coefficients
```

 Z_{ij} free-space mutual impedance between strip vee dipole i and strip vee dipole j α_n, β_n direction angles of segment n λ free-space wavelength ϕ reference angle pitch-plane reference angle

 ϕ_n roll-plane reference angle

CONFIGURATION

An annular slot antenna having an inner diameter of 48.37 cm and an outer diameter of 58.31 cm and radiating over the frequency range of 150 to 400 MHz is to be considered for operation at the points A, B₁, and B₂, as indicated in figure 1 on a diagram of the Space Shuttle orbiter. The roll and pitch planes in which radiation patterns are to be determined, as well as the roll-plane reference angle ϕ_r and the pitch-plane reference angle ϕ_p , are also defined in figure 1. Roll-plane patterns are to be obtained for the annular slot excited at A (designated as case 1 throughout this paper) and at B₁ (case 2), and simultaneously excited at B₁ and B₂ (case 3). Pitch-plane patterns are also to be obtained for the annular slot excited at A (case 4).

Since the orbiter is electrically large over the required operating frequency range, a reasonable approximation is to assume the roll-plane pattern to be the pattern produced by an annular slot located on an infinitely long, perfectly conducting cylinder of the same cross section as the cross section of the orbiter in the plane containing the points A, B₁, and B₂ and perpendicular to the longitudinal axis of the orbiter. Similarly, it is reasonable to assume the pitch-plane pattern to be the pattern produced by an annular slot located on an infinitely long, perfectly conducting cylinder having the same cross section as the cross section of the orbiter in the plane containing point A and the longitudinal axis of the orbiter. However, since the size of the orbiter in the pitch plane exceeds the limitations of RIEF, a foreshortened approximation (to be explained later) is used for the pitch-plane cross section in order to make pattern computations.

INTEGRAL EQUATION FORMULATION

Richmond (ref. 5) has recently developed a Galerkin's method IEF solution (RIEF) and a user-oriented digital computer program for computing the radiation pattern of one or more axial slot antennas mounted on a conducting cylinder of infinite length and arbitrary cross section. The slot antennas may be either narrow (with infinitesimal width) or wide (with finite width). For simplicity, application of RIEF in this paper is for several narrow axial slots mounted on a perfectly conducting cylinder, the geometry of which is shown in figure 2. The electromagnetic field is assumed to be transverse electric (TE) to the cylinder axis; thus, the radiation field consists of only the E_{ϕ} component. A synopsis of RIEF is given in this paper; further details are given in reference 5.

In RIEF, the cylinder cross-section profile is initially subdivided into a specified number N of straight segments, as shown in figure 2. The point of intersection P_n (n = 1, 2, . . ., N) of any two adjacent segments 3 n and n + 1 is regarded as a "delta gap" across which a known voltage V_n may exist for excitation of the structure. Points for which $V_n \neq 0$ are interpreted as narrow axial slots so that in the subdivision process any narrow axial slot on the cylinder is required to be located at a point of intersection.

Segment n (n = 1, 2, . . ., N) is characterized by the end points $P_n = (x_n, y_n)$ and $P_{n+1} = (x_{n+1}, y_{n+1})$, segment length $d_n = \left[(x_{n+1} - x_n)^2 + (y_{n+1} - y_n)^2 \right]^{1/2}$, and orientation unit vector $\hat{s}_n = \cos \alpha_n \hat{x} + \cos \beta_n \hat{y}$ directed from P_n to P_{n+1} , where $\cos \alpha_n = (x_{n+1} - x_n)/d_n$ and $\cos \beta_n = (y_{n+1} - y_n)/d_n$. For accurate radiation pattern computations, all segments should be no greater than $\lambda/4$ in length where λ is the free-space wavelength. Accuracy is enhanced, however, if the length is moderately shorter than $\lambda/4$, especially for segments close to a narrow axial slot.

Any two adjacent segments represent the arms of a strip vee dipole; in particular, segments n-1 and n constitute strip vee dipole n, as indicated in figure 2. In all, there are N overlapping strip vee dipoles. A useful representation for the surface current distribution on dipole n is the piecewise sinusoidal representation

$$\overline{J}_{Dn}(t) = I_n \frac{\sin kt_{n-1}}{\sin kd_{n-1}} \hat{s}_{n-1} \qquad (0 \le t_{n-1} \le d_{n-1}) \qquad (1a)$$

$$\overline{J}_{Dn}(t) = I_n \frac{\sin k(d_n - t_n)}{\sin kd_n} \hat{s}_n \qquad (0 \le t_n \le d_n)$$
 (1b)

³Since the array of segments in figure 2 is a closed array of segments, the index n+1 must be replaced by 1 for n=N, and the index n-1 must be replaced by N for n=1.

where t_{n-1} is the distance from P_{n-1} along segment n-1 and t_n is the distance from P_n along segment n, and where $k=2\pi/\lambda$. From equations (1), the resultant surface current distribution on segment n is given by

$$\overline{J}_{n}(t) = \frac{\left[I_{n+1} \sin kt_{n} + I_{n} \sin k(d_{n} - t_{n})\right]}{\sin kd_{n}} \hat{s}_{n} \qquad (0 \le t_{n} \le d_{n})$$
(2)

where t_n is the distance from P_n along segment n.

The end point surface current densities I_n (n = 1, 2, . . ., N) which are unknown quantities are determined by solving the system of N simultaneous equations

$$[Z][I] = [V] \tag{3}$$

where [Z] is a N × N impedance coefficient matrix which can be determined (ref. 5), [I] is the N × 1 (column) matrix of unknown dipole mode currents, and [V] is the N × 1 (column) excitation matrix, the elements of which are nonzero only for actual slots. The Z_{ij} (i,j = 1, 2, . . ., N) in [Z] physically represent the free-space mutual impedance between strip vee dipole i and strip vee dipole j under the condition for which the current distribution on each dipole is piecewise sinusoidal. Thus,

$$Z_{ij} = -\frac{1}{I_i I_j} \left[\int_0^{d_{j-1}} \overline{E}_i(t_{j-1}) \cdot \overline{J}_{Dj}(t_{j-1}) dt_{j-1} + \int_0^{d_j} \overline{E}_i(t_j) \cdot \overline{J}_{Dj}(t_j) dt_j \right]$$
(4)

where \overline{E}_i is the electric field radiated by dipole i due to the current distribution \overline{J}_{Di} given by equations (1) and evaluated on the surface of dipole j (the surface of and points interior to dipole j are assumed to be replaced by free space), \overline{J}_{Dj} is also given by equations (1), and t_j is the distance from P_{j-1} along segment j. Details for computing Z_{ij} are given in reference 5.

The solution for [I] yields the coefficients I_n needed in equation (2) for the computation of the current distribution on the cylinder. Standard techniques are then employed to compute the radiation field E_{ϕ} . Patterns are computed with the coordinate origin 0 as the phase center.

REPRESENTATION OF AN ANNULAR SLOT

In order for the roll- and pitch-plane patterns of the annular slot in figure 1 to be computed, subject to the approximations given, the radiation pattern of a thin annular slot located on an infinitely long, perfectly conducting, noncircular cylinder must be computed in a plane perpendicular to the axis of the cylinder and containing the center of the

annular slot, as shown in figure 3. Under the assumption that the field in the annular slot aperture is transverse electric and magnetic (TEM) to the aperture, \overline{E} on any radial line of the slot aperture is equal in amplitude and opposite in phase to \overline{E} on the diametrically opposite radial line. Consequently, application of RIEF for narrow axial slots requires that the annular slot be simulated, as shown in figure 3, by an equivalent array of two infinitely long, antiphase narrow axial slots of equal amplitude, and spaced by the distance $d_M = (d_I + d_O)/2$, where d_I and d_O are the inner and outer diameters, respectively, of the annular slot.

EXPERIMENTAL MODELS

Two 1/35-scale models were constructed for obtaining radiation patterns experimentally. The first model, shown in figure 4, is a cylindrical model having a cross section similar to the cross section of the orbiter in the roll plane. The second model, shown in figure 5, is a three-dimensional model with the side and front views clearly showing the pitch- and roll-plane cross-section profiles, respectively. For the scale models, the annular slot inner diameter scales to 1.382 cm and the outer diameter scales to 1.666 cm, and the frequency range⁴ of 150 to 400 MHz scales to 5.250 to 14.000 GHz. For each model, the location of the annular slot antenna corresponds to point A of figure 1; consequently, the cylindrical model may be used to obtain radiation patterns only for cases 1 and 4.

Since RIEF is directly applicable only for two-dimensional models, the purpose of the cylindrical model is to verify RIEF experimentally. The infinitely long cylinder assumed in RIEF is simulated experimentally by a cylinder of length 39.04 cm, as shown in figure 4. This length is 6.83 wavelengths at the lowest desired scaled frequency of 5.250 GHz.

NUMERICAL COMPUTATIONS AND EXPERIMENT

RIEF was applied to obtain computations of the radiation patterns for cases 1 to 4. Computations were made over the scaled frequency range for the 1/35-scale model. For each pattern computation, only a coarse number N' ($N' \leq N+1$) of points were initially specified. The remaining points P_n , particularly those in the shadow region of the slot(s), were automatically selected with the aid of a spline curve fit procedure. This procedure not only reduces the number of input data points but also is desirable in order

 $^{^4}$ In this paper, frequencies given in GHz refer to the 1/35-scale models; and those given in MHz refer to the full-scale structure.

that a smooth profile be obtained, since in most cases the points must be read graphically from drawings and inaccuracies in profile smoothness often result.

The digital computer programs which were used to obtain radiation pattern computations are given in appendix A. An example illustrating the use of the programs is given in appendix B.

The upper limit of the perimeter of a cylinder which can be analyzed with RIEF is dependent on the storage capacity of the computer used to obtain computations, which, in this paper, were made on a Control Data Corporation 6600 computer at NASA Langley Research Center. It was found that for this computer and for the auxiliary plotting programs employed, the maximum number of simultaneous equations which could be solved without exceeding storage capacity was 180. Under this condition, the upper limit of the perimeter is from 30 to 35 wavelengths, and the running time for a typical pattern (10 increments) is about 100 seconds.

Roll-Plane Radiation Patterns

The roll-plane cross section in the plane containing the points A, B_1 , and B_2 is described for case 1 by 154 points shown in figure 6. When these points are connected, a polygon of perimeter 70.58 cm is formed, and 154 simultaneous equations are solved to compute the radiation pattern.

The geometry for cases 2 and 3 is shown in figures 7 and 8, respectively. The point locations in figures 7 and 8 are similar to the point locations in figure 6, except that in figures 7 and 8 additional points were added in the vicinity of the sources. This procedure is necessary to obtain more reliable data for cases 2 and 3. The perimeter, number of points and simultaneous equations, the scaled coordinates of the center of the annular slots, and the scaled coordinates and voltage strengths of the equivalent axial slots are conveniently presented for each case in table I.

Experimental verification for the roll-plane patterns was achieved with measurements made on both experimental models at 10.900 GHz (311.4 MHz). The results are given in figure 9 where a comparison with theory is given. Computed radiation patterns for cases 1 to 3 over the range of 5.250 to 13.125 GHz (150 to 375 MHz) are presented in figure 10.

Pitch-Plane Radiation Patterns

Computations of the pitch-plane patterns (case 4) are more difficult, since the perimeter of the pitch-plane cross-section profile varies roughly from 60 to 150 wavelengths over the desired frequency range. Under this condition, application of RIEF would require the solution of a minimum of 300 simultaneous equations, a number which leads to

excessive computer storage requirements. The profile electrical size would indicate GTD to be a more useful approach for this computation. Unfortunately, GTD is not easily applicable to this profile, since practical use of GTD is limited currently to profiles convex in shape.

However, since patterns are of interest primarily in the lit region and in the forward direction, the portion of the pitch-plane cross-section profile from the back of the cockpit to the tail may be ignored by closing the cross section, as indicated in figure 11. The result is a polygon described by 178 points, leading to a system of 178 simultaneous equations. (See table I.) A computed pitch-plane pattern for the geometry of figure 11 is given in figure 12, with experimental verification obtained from the three-dimensional model of figure 5.

DISCUSSION OF RESULTS

The reliability of computing principal-plane radiation patterns of thin annular slots mounted on a spacecraft, such as the Space Shuttle orbiter, by use of RIEF and the approximations made is enhanced by the excellent agreement between theory and experiment shown in figures 9 and 12. It is evident from figures 9, 10, and 12 that a null occurs in the lit region on the normal to the plane of the annular slot aperture, and that major lobes occur at roughly 60° on either side of the normal. This result is generally true for a thin annular slot of mean radius a mounted on a ground plane of infinite extent and having a value of ka ≤ 2 , pattern breakup occurring for larger values 5 of ka. (See ref. 18, pp. 8-8 and 8-9.)

If the annular slot is located at A, figures 9 and 10(a) reveal that ground coverage is poor, especially in the shadow regions. An improvement in ground coverage is achieved, as indicated in figure 10(b), if the annular slot is located at B_1 (or B_2), but coverage is asymmetrical unless both locations and a switching arrangement are used. However, symmetry is achieved if the annular slots at B_1 and B_2 are excited simultaneously, as indicated in figure 10(c), but pattern breakup is evident, as is generally encountered for arrays of widely separated antennas on electrically large bodies. Obviously, excellent ground coverage would be achieved if an annular slot were located on the underside of the orbiter, but this location is undesirable because of the excessive heat anticipated during atmospheric reentry. Figure 10 reveals that not only is the ripple structure finer as the frequency is increased, but also the fields in the shadow region are relatively weaker; that is, shadow boundaries are more sharply defined. Finally, figure 12 reveals that desired forward direction coverage results if the annular slot is located at A.

 $^{^5 \}text{Maximum}$ value of $\text{ ka} = \text{k}(\text{d}_M/2)$ for the annular slot in this paper is 2.25 at a frequency of 400 MHz.

It was noted that for all cases the phase of the radiation field fluctuates through many cycles, particularly in the shadow regions, as either $\phi_{\mathbf{r}}$ or $\phi_{\mathbf{p}}$ is varied from 0^{O} to 360^{O} . This result is generally true for antennas on electrically large bodies.

CONCLUDING REMARKS

In this paper, a digital computer program based on a two-dimensional integral equation formulation (Richmond Integral Equation Formulation) was applied to predict radiation patterns of annular slot antennas mounted on the Space Shuttle orbiter. The excellent agreement between computed patterns and patterns obtained from measurements made on 1/35-scale models supports the use of this approach for future similar spacecraft or aircraft antenna design problems.

By use of the Richmond Integral Equation Formulation, additional radiation patterns may be predicted easily as the design parameters of the orbiter or antenna are changed; thus, it is insured that proper radiation pattern coverage is maintained from design to construction. The Richmond Integral Equation Formulation may also be readily applied to predict patterns of other antennas of interest such as slot antennas. If antenna arrays are considered, the solution to the set of simultaneous equations automatically includes mutual coupling effects.

Although limited by computer storage requirements, the Richmond Integral Equation Formulation is characterized by ease of input data points describing the profile and antenna to be analyzed. In contrast, the use of geometrical theory of diffraction requires a much more critical selection of input data points and equations describing the body surface between any successive pair of points, since smooth functions of the radius of curvature and its arc length derivatives are required.

Langley Research Center,

National Aeronautics and Space Administration,

Hampton, Va., March 25, 1974.

APPENDIX A

DIGITAL COMPUTER PROGRAMS

The purpose of this appendix is to present and to describe the use of the digital computer programs employed to obtain computations and plots of the radiation patterns for the Space Shuttle annular slot antennas described in the text. As indicated in the text, the pattern computations and plots obtained by use of RIEF (see fig. 2) require a knowledge of (a) the X- and Y-coordinates of the end points of the segments used to approximate the cross-sectional profile of the orbiter in the plane in which the pattern is to be computed, and (b) the voltage strengths and the point indices of the equivalent narrow axial slot antennas representing the annular slot antennas. It should be noted in figure 2 that the end points of the profile segments are indexed in a counterclockwise direction, the first point being located at $\phi = 0^{\circ}$.

It is also indicated in the text that only a coarse number of points are specified initially and that the remaining points are generated with the aid of a spline fit procedure. In this appendix and in appendix B, for the sake of nomenclature, an initially specified point is designated as a "coarse point," and the index of the coarse point as the "coarse point index." Any one of the end points generated by the spline curve fit procedure and required for the computation of a radiation pattern is designated as an "actual point" and the index as the "actual point index."

The N' coarse points are defined and are indexed in the same manner as the N actual points defined in figure 2. However, coarse point N' is located and specified at the same point as coarse point 1. After the spline fit procedure is applied to generate the N actual points from specification of the coarse points, it will be found that the actual points consist of all the coarse points (except coarse point N') and additional points located between any two consecutive coarse points. If it is desired that the coarse points and the actual points be identical, then N' = N + 1. Under this condition, no points are generated between any pair of consecutive coarse points. If the spline fit procedure is used to generate additional points, however, then $N' \leq N + 1$.

The main program and all supporting subroutines used for analysis of the Space Shuttle annular slot antennas are given at the end of this appendix. This program requires a specification of the following data: (a) number of coarse points N' and the X- and Y-coordinates of each coarse point, (b) the indices of the coarse points at which the equivalent narrow axial slots are located, 6 (c) the voltage strengths of the narrow axial slots,

⁶A narrow axial slot is required to be located at a coarse point; conversely, a coarse point is required to be designated at a narrow axial slot.

(d) the initial frequency, the incremental frequency, and the final frequency of the frequency range over which radiation patterns are to be computed, (e) an array of integers which indicate the number of segments that are to be generated between any two consecutive coarse points by means of a two-dimensional spline fit method, and (f) the coarse point index at which the spline fit is to be initiated.⁷

After these data are specified, the main program calls subroutine SPLFIT, which, in turn, calls subroutine SPFIT2, to generate the X- and Y-coordinates of the N actual points from the spline fit procedure. The main program then calls subroutine CVRTV to convert the coarse point indices of the narrow axial slots to the actual point indices as defined in figure 2. After calls to subroutines SPLFIT and CVRTV are made, then the main program, as a function of frequency, converts the X- and Y-coordinates of the N actual points from units of centimeters to units of wavelength and calls subroutine TESLOT to compute the radiation pattern.

Subroutine SKETCH is a subroutine which generates instructions on a magnetic tape used to drive the CalComp (California Computer Products, Inc.) plotter. Called by the main program, subroutine SKETCH plots the set of N actual points describing the cross-section profile. A call to subroutine SKETCH produces a plot similar to the plots given in figures 6 to 8, and immediately below the plot a scale is given. For the first call to subroutine SKETCH, the X- and Y-coordinates are in centimeters, and the scale is in centimeters per inch (on the plotting paper). For the remaining calls to subroutine SKETCH, the X- and Y-coordinates are in wavelengths, and the scale is in wavelengths per inch (on the plotting paper).

Subroutine DBPLOT is a subroutine also used to generate instructions on a magnetic tape for driving the CalComp plotter. At each frequency over the range of frequencies for which a radiation pattern is desired, subroutine DBPLOT produces a polar plot of radiation field magnitude, in decibels, against ϕ_p or ϕ_r and a rectangular plot of radiation field phase, in degrees, against ϕ_p or ϕ_r .

The programs given apply for obtaining computations and plots of radiation patterns of, in general, a set of narrow axial slot antennas (or annular slot antennas, each of which is represented by a pair of narrow axial slots) mounted on a cylinder of arbitrary cross section. In order to use the programs, the user is required to have additional knowledge of how the input data are set up for transmission to the main program and to have a knowledge of how the arrays in the main program are dimensioned.

⁷The point at which the spline fit is initiated is chosen to be in an approximately linear region of the profile.

Input Data

All input data are read in through the main program. The first input data card contains the following quantities: NPTIN, ISTART, NPORT, MADM, KI, KWRT1, KWRT2, KWRT3, ISKIPP, ISIZEP, HPAT, FMCO, FMCD, and FMCF. The quantities are defined as follows:

- NPTIN Number of coarse points describing the cross-section profile (NPTIN \leq N + 1, where N = Actual number of points, as defined in fig. 2)
- ISTART Coarse point index at which the spline fit is to be initiated $(1 \le ISTART \le NPTIN)$
- NPORT Number of narrow axial slots (1 ≤ NPORT ≤ NPTIN)
- MADM 1 or 0. For MADM = 1, the short-circuit admittance matrix for all ports (that is, narrow axial slots) is computed. For MADM = 0, this computation is avoided. (In this paper, MADM = 0 for all computations)
- KI number of angular values of ϕ_p or ϕ_r at which the radiation field E_ϕ is computed (KI = 361, which is used for all computations in this paper, gives pattern computations in 1^0 increments)
- KWRT1 1 or 0. KWRT1 = 1 gives a write out of the X- and Y-coordinates, in wavelengths, of the N actual points and the distance, in wavelengths, between any two consecutive segments. KWRT1 = 0 avoids this write out
- KWRT2 1 or 0. KWRT2 = 1 gives a write out of the real part, the imaginary part, the magnitude, and the phase of the radiation field E_{ϕ} for each of the KI angular values of $\phi_{\mathbf{p}}$ or $\phi_{\mathbf{r}}$. KWRT2 = 0 avoids this write out
- KWRT3 = 1 gives a write out of the value of E_{ϕ} in decibels, the normalized value of E_{ϕ} , and the phase of the radiation field E_{ϕ} for each of the KI angular values of ϕ_p or ϕ_r . The value of E_{ϕ} in decibels and the normalized value of E_{ϕ} are with respect to the maximum value of E_{ϕ} occurring among the KI values of E_{ϕ} computed. KWRT3 = 0 avoids this write out
- ISKIPP Integer required for obtaining radiation pattern plots. ISKIPP is positive for line and symbol plots, and is negative for symbol-only plots. The magnitude of ISKIPP specifies the alternate number of data points at which a symbol is plotted

ISIZEP 1, 2, or 3. ISIZEP = 1 produces small size symbols, ISIZEP = 2 produces medium size symbols, and ISIZEP = 3 produces large size symbols

HPAT A floating point number that is five times the radius, in inches, of the polar plots to be made with subroutine DBPLOT.

FMCO Initial frequency, in MHz, for which pattern computations are to be made

FMCD Incremental frequency, in MHz, for which pattern computations are to be made

FMCF Final frequency, in MHz, for which pattern computations are to be made

After the first data card is read in, a set of NPORT data cards is then read into the main program. On each card are specified the port index I, the coarse point index JVGS(I), and the complex voltage strength VGS(I) of the Ith port (that is, the Ith narrow axial slot).

Next, a set of NPTIN data cards is read into the main program. On each card is an identifying integer IGNORE, the X-coordinate PNTIN (I,1), in cm, and the Y-coordinate PNTIN (I,2), in cm, of the Ith coarse point.

Finally, the IDIVD array of integers is read into the main program. The IDIVD array of integers is a set of NSEG = (NPTIN - 1) integers INDIVID(I) (I = 1, 2, . . ., NSEG) which specify the number of segments that will occur between any pair of consecutive coarse points as a result of the use of subroutines SPLFIT and SPFIT2. Specifically, if IDIVD(I) = IX, then IX segments will occur (or IX-1 points will be generated) between coarse point I and coarse point I + 1. All segments between any two consecutive coarse points are approximately equal in length, which should be no greater than $\lambda/4$ at the highest frequency for which radiation pattern computations are desired. It should be noted that the sum of the NSEG integers in the IDIVD array is equal to the number of actual segments and points . N generated for use in subroutine TESLOT. As noted in the text, this number should not exceed 180 if the Control Data Corporation 6600 computer at NASA Langley Research Center is employed. If the generation of spline-fitted points is desired to be avoided altogether (that is, the coarse points and the actual points are identical. except that coarse point N' is omitted), it is necessary only to set all NSEG integers equal to 1 in the IDIVD array. Under this condition, the N + 1 coarse points become the N points required by subroutine TESLOT for computation of the radiation pattern.

Dimension of Arrays in the Main Program

In some applications, the user may wish to change the dimensions of the arrays in the main program. To change the dimensions, define the following integers:

n₁ maximum number of actual points (and simultaneous equations)

n₂ maximum number of narrow axial slots

ng maximum number of angular values for which radiation patterns are desired

 $n_4 \ge n_2 + 2$

n₅ maximum number of coarse points

Then the following statements must appear in the main program:

PROGRAM MAIN (INPUT, OUTPUT, . . .)

COMMON/BLK1/KNTPLT, XNEW, YNEW

COMPLEX VGS, C, CJ, EPP, YMMHO, ZINPUT

DIMENSION $X(n_1)$, $Y(n_1)$, $XC(n_1)$, $YC(n_1)$, $VGS(n_2)$, $IVGS(n_2)$,

1 $D(n_1)$, $C(n_1,n_1)$, $CJ(n_1)$, $EPP(n_1)$, $YMMHO(n_2,n_2)$,

2 THTA (n_4) , AEPH1 (n_4) , RATIO (n_4) , PEPH1 (n_4) , AEY (n_4) ,

3 $XW(n_1)$, $YW(n_1)$, $PNTIN(n_5,2)$, $SPNTIN(n_5,2)$, $IDIVD(n_5)$,

4 ELEN(n_5), COEF(n_5 ,4,2), XPNT(n_1), YPNT(n_1), JVGS(n_2)

 $MAXP = n_5$

 $MAXCO = n_5$

 $MXNPT = n_1$

 $MXPORT = n_2$

CALL CALCOMP

CALL LEROY

KNTPLT = 0

END

For the main program n_1 corresponds to 170, n_2 to 4, n_3 to 723, n_4 to 725, and n_5 to 57. A restriction is that the sum of the NSEG integers in the IDIVD array must not exceed the value of n_1 .

Main Computer Program for Obtaining Computations of Space Shuttle Annular Slot Antenna Radiation Patterns

```
PRUGRAM MAINLINPUT, OUTPUT, TAPES=INPUT, TAPE6=OUTPUT)
C.
      J. EARL JONES, ARS-TRB-FIL, PHUNE 3631, MAIL STOP 490--NASA45897C
      С
       PURPOSE TO SET UP INPUT DATA TO SUBROUTINES TESLOT AND OBPLOT
                 TO OBTAIN. OVER A RANGE OF FREQUENCIES. RADIATION
                 PATTERN CALCULATIONS AND PLOTS AND TO OBTAIN THE
Ċ
                 SHORT CIRCUIT AUMITTANCE MATRIX OF AN ARRAY OF TE
                 SLOTS ON A PERFECTLY CONDUCTING CYLINDER OF ARBITRARY
Ċ
                 CROSS SECTION
C
                     FMCO=INITIAL FREQUENCY. MHZ.
       DEFINITIONS
C
                     FMCD=INCREMENTAL FREQUENCY. MHZ.
                     FMCF=FINAL FREQUENCY, MHZ.
С
C
C
                     NPOINT, MANPI, NPORT, MAPORT, MADM, KI, KWRT1, KWRT2,
Ç
                     KWRT3, VGS, IVGS, AW, YW, THTA, AEPHI, RATIC, PEPHI, YMMHO, *
                     ZINPUT. AND LPTEPH ARE DEFINED IN SUBROUTINE
C
                     TESLOT. ISKIPP, ISTZEP, AND HPAT ARE DEFINED IN
¢
                     SUBROUTINE DEPLOT.
C
                     IF ANALYSIS OCCURS OVER A RANGE OF FREQUENCIES.
Ĺ
                     VALUES IN XW AND YW ARE INITIALLY READ IN IN CM.
Ċ.
                     AND THESE VALUES ARE STORED IN AC AND YO ARRAYS.
C
                     IF EITHER FACL, FACU, OR FACE ARE SET TO A
                     NEGATIVE VALUE, THEN VALUES IN XW AND YW ARE
C
                     UNUERSTOUD TO BE READ IN IN WAVELENGTHS.
C
      COMMON /BEK1/ KNTPLT.XNEW.YNEW
      COMPLEX VGS, C, CJ, EPP, YMMHO, ZINPUT
     UIMENSION X(170), Y(170), XC(170), YC(170), VGS(4), IVGS(4).
     1 L(176), C(170, 176), CJ(176), EPP(170), YMMHC(4,4),
     Z THTA(725), 46PH1(725), RATIG(725), PEPH1(725), AEY(725),
     3 XW(170),YW(17C),PNTIN(57,2),SPNTIN(57,2),ID(VD(57),
     4 ELEN(37), COEF (57,4,2), XPNT(170), YFNT(170), JVGS(4)
     MAXP=57
     MAXCU=57
      MXNPT = 170
     MXPDRT=4
     CALL CALCOMP
     CALL LERGY
     KNTPLT=0
     KTV = 1
   i REAU(5,131) NPTIN, ISTART, NPURT, MADM, KI, KHPT1, KWRT2, KWRT3,
    1 ISKIPP, ISIZEP, HPAT, FMCD, FMCD, FMCF
      IF(EUF,5) 4,6
   4 IF(KNTPLT.EQ.O) GO TO 5
     CALL CALPET(XNEW, YNEW, -999)
   5 STOP
```

```
6 CONTINUE
   IF((FMCO.LE.O.).CR.(FMCO.LE.O.).OR.(FMCF.LE.O.)) KTV=0
   REAU(5,103) (I,JVGS(I),VGS(I),I=1,NPORT)
   READ(5,104) (IGNURE, PATIN(1,1), PATIN(1,2), I=1, NPTIN)
   NSEG=NPTIN+1
   READ(5,108) (IDIVD(I),I=1,NSEG)
   CALL SPLEIT (MAXP. MAXCC. NPT IN. ISTART, PNT IN. IDIVD. ELEN,
  1 COEF. SPNTIN. XPAT. YPNT. APCINT. XM. YW)
   CALL CVRTV(NPTIN, NPDRT, JVGS, IDIVD, IVGS)
   wRITE(6.2001
   WRITE(6.204) NPTIN.ISTART
   WRITE (6,208)
   wRITE(6,206)
   wRITE(6.108) (IBIVD(I).I=1.NSEG)
   WRITE(6.208)
   WRITE(6,201)
   WRITE(0,151) NPCINT, MXNPT, NPCKT, MXPORT, MADM, KI, KWRT1,
  1 KWRT2.KWRT3.ISKIPP.ISIZEP.HPAT.FMCD.FMCD.FMCD.
   WRITE(6.208)
   IF(KTV.EQ.O) IUNITS=2
   IF(KTV.EQ.1) | IUNITS=1
   CALL SKETCH(xw, Yw, NPOINT, HPAT, 1, 22, 1, IUNITS)
   IF(KTV-EQ-0) GC TO 14
   UO 10 I=1.NPOINT
   xC(I) = xw(I)
   YC([]=YW([]
10 CONTINUE
   WRITE(6,209)
   wRITE(6,216)
   wRITE(6,152) (1,xC(1),YC(1),I=1,NPOINT)
   WRITE(5,2001
   HRITE16,2021
   FMC=FMCO
11 WVL = 30000 - /FMC
   WRITE(6,211) FMC.WVL
   UU 12 I=1, NPOINT
   XWLI] = XC(I)/WYL
   YW(1)=YC(I)/WVL
12 CONTINUE
   WRITE(6,208)
   CALL SKETCH(XW,YW,NPCINT,HPAT,1,22,1,2)
14 CUNTINUE
   CALL TESLOT (NPCINT, MANPT, NPORT, MXPORT, MACM, KI.
  1 KWRT1, KWRT2, KWRT3,
  AWIYWIVGSILVGSIXIYILICICIJEFFITHTA, AEPHIIRATIO, PEPHII
  3 YMMHO, ZIMPUT, LPTEPH)
   IF(LPTePH.EQ.O) GO TO 18
   CALL OBPLOT (HPAT, KI, ISKIPP, ISIZEP, THTA, AEPH1, AEY, PEPH1)
   GC TU 20
18 WRITE(5.224) LPTEPH
20 CONTINUE
   WRITE (5,2001
   WRITE (5,218)
   wRITE(6.219)
   00 24 I=1.NPOP*
   DC 24 J=1.NPCRT
   ISCRPT=IVGS(I)
   JSCRPT=IVGS(J)
   WRITE(6,154) ISCRPT, I, USCRPT, J, YMMHC(I, J)
24 CONTINUE
   IH(NPORT-UT-1) GG TO 28
   WRITE(U,208)
   WHITE (6,222) ZINPUT
```

18

1

1

ì

```
29 [F(KTV.EQ.O) GC TO 1
   FMC=FMC+FMCD
   wRTTF (6 + 202)
   IF(FMC.LE.FMCF) GG TO 11
   60 Tu 1
101 FURMAT(1015.F5.0.3F7.C)
103 FORMAT(215,2F10.0)
104 FORMAT(IS.2F10.C)
108 FORMAT(4012)
151 FULMAT(1117.F7.2.3F7.1)
152 FOR MAT(110, 2F15,4)
154 FURMAT(112,317,2F16.8)
200 FORMAT(110H *******************************
  201 FORMAT(105H NPOINT MANPT NPCRT MAPORT MADM K! KWRT1 KWRT
  12 KWRT3 ISKIPP ISIZEP HPAT FMCO FMCF)
1-----//
204 FORMAT(1x.6HNPTIN=12.3x.7HISTART=12./)
206 FORMATIIX, 18HIDIVD ARRAY VALUES, //
208 FÜRMAT(/)
209 FORMAT(13x, 18HLECATION OF PCINTS)
210 FORMAT(5X,35HPCINT
                        X. CM.
                                     Y. CM. //
211 FOF MATISX, 10HF REQUENCY = F7. 1,4H MHZ, /,5X,11HWAVELENGTH = F7. 2,4H CM.)
218 FORMAT(5x, 31HSHCRT-CIRCUIT ADMITTANCE MATRIX, /)
219 FORMATISX.60H PCINT PURT PUINT PORT
                                       G. MILLIMHOS
  1LLIMHOS)
222 FORMAT(5X, 8HZINPUT=(F16.8, 1H, F16.8, 6H) CHMS, /)
224 FORMAT(5X.7HLPTEPH=12.14H FOR THIS CASE./)
```

Subroutine TESLOT

SUBROUTINE TESLET (NPCINT, MXNPT, NPORT, MXPORT, MADM, KI.

```
1 KARTI . KWET2 . KWET3 .
     Z XW+YW+VGS+IVGS+X+Y+D+C+CJ+EPP+THTA+AEPH1+RATIO+PEPH1+
     3 YMMHO.ZINPUT.LPTEPH)
C
      ¢
      * PURPOSE TO CALCULATE THE RADIATION PATTERN AND TO CALCULATE
C
                THE SHORT CIRCUIT AUMITTANCE MATRIX OF AN ARRAY OF TE *
C
C
                SLOT ANTENNAS ECCATED ON A PERFECTLY CONDUCTING
C
                CYLINDER OF ARBITRARY CROSS SECTION
C
C
                   NPUINT=NUMBER OF POINTS DESCRIBING THE CYLINDER
C
                   MANPT=MAXIMUM PERMISSIBLE NUMBER OF POINTS
r
                   NPORT=NUMBER OF SLOT ANTENNAS IN THE ARRAY
                   MXPORT=MAXIMUM PERMISSIBLE NUMBER OF SLOT ANTENNAS #
ť.
(
     *
                   MACM=1 IF SHORT CIRCUIT AUMITTANCE MATRIX IS TO BE *
     *
                        COMPUTED, O FOR NO COMPUTATIONS. MATRIX IS
(
     *
                        ALWAYS COMPUTED IF NPORT=1, HOWEVER.
                   KI=NUMBER OF RADIATION PATTERN POINTS TO BE COMPU-
С
C
      #
                      TED/ PLUTTED PER CASE. FOR 5 DEG. INCREMENTS
                      SET KI=72+1, FOR 4 DEG. INCREMENTS SET KI=9C+1.
C
Ü
                      FOR 3 DEG. INCREMENTS SET KI=120+1, ETC.
Ü
                   KHRTI=1 FOR WRITEUUT OF ARPAY OF POINT LOCATIONS.
C
                         AND ARRAY OF SLOT ANTENNA LOCATIONS.
ŗ
                         C FOR NO WRITEOUT
                   KWRT2=1 FOR WRITEBUT OF RELATIVE RACIATION PATTERN, *
С
                         O FUR NU WRITEDUT
```

```
KWRIBEL FOR WRITEGUT OF US RADIATION PATTERN.
(
                          S FOR NO WRITEDUT
ί.
                    XWEARRAY OF X-VALUES OF CCCRDINATES. WAVELENGTHS
ſ
                    YWEARRAY OF Y-VALUES OF COCRDINATES. WAVELENGTHS
Ċ
                    VISSARRAY OF VALUES OF SLOT VOLTAGES
C
                    TVGS=ARRAY OF VALUES OF THE LOCATIONS (CORRESPON-
Ĺ
                         DING TO THOSE IN THE XW AND YW ARRAYS) OF THE
Ċ.
                         SLCT VOLTAGES IN THE VGS ARRAY
C
c
       BUT PUT DATA THIA=ARKAY CONTAINING THE ANGULAR VALUES
€
                          (LEGREES) FOR WHICH THE RADIATION FIELD IS
C.
                          TO BE PLUTTED. THTA(1)=0., THTA(KI)=360.
      +
L
                     AFPHI=ARRAY OF MAGNITUDE VALUES (OB) CORRESPONDING*
C
                           TO ANGULAR VALUES IN THIA ARRAY
C
C.
                      PATID=ARRAY OF VALUES OF (E/EMAX) CORRESPONDING
                            TO ANGULAR VALUES IN THTA ARRAY
C
                      PEPHIEARKAY WE PHASE VALUES (DEGREES)
٢
                            CCKRESPUNDING TO ANGULAR VALUES IN THIA
\mathbf{C}
C
                            ARR AY
                      YMMHO=SHORT CIRCUIT ADMITTANCE MATRIX, MILLIMHOS
      *
C
ď
                      ZINPUT=INPUT IMPEDANCE, CHMS (CNLY FOR NPORT=1)
                     LPTEPH=1 INDICATES A RADIATION FIELD PLCT FOR USE
C
                             IN DEPLOT WAS COMPUTED. O INDICATES THERE
Ċ
                             WAS NO RADIATED FIELD. AND HENCE NO PLOT
C
C
                             WAS COMPUTED
ť.
        K,Y,D,C,CJ,EPP ARE ARKAYS USED FOR INTERMEDIATE CALCULATIONS
C
C
      * RESTRICTIONS IN CALLING PROGRAM, VGS.C.C.J.EPP.YMMHC.AND ZINPUT*
ť.
                       MUST BE TYPLE CUMPLEX. ARRAYS XW.YW.X.Y.D.EPP.
                       AND CU MUST BE DIMENSIONED WITH THE VALUE OF
      *
C
                       MXPNT AND C MUST BE DIMENSIONED AS CEMXPNT.MXPNT.*
C
C
                       IN CALLING PROGRAM. ARPAYS VGS AND IVGS MUST
C
                       BE DIMENSIONED WITH THE VALUE OF MAPORT AND YMMHO*
                       MUST BE DIMENSIONED AS YMMHE (MXPCRT, MXPORT) IN
¢
C
                       CALLING PROGRAM. FOR PLOT PURPOSES. ARRAYS THTA.*
C
                       AEPHI, RATIC, AND PEPHI MUST BE DIMENSIONED AT
                       LEAST AS LARGE AS THE VALUE (KI+2) IN CALLING
C
C
                       PROGRAM.
C
      CCMPLEX VGS, YMMHO, CJ. C. EPP. CLT. EPS. ZINPLT
      COMPLEX P11, P12, C11, U12, Q21, U22, EP1, EP2
      DIMENSION X(1), Y(1), U(1), VGS(1), IVGS(1), C(MXNPT.1).
     1 CJ(1), YMMHO(MXPCRT, 1), EPP(1), AEPH1(1), RATIO(1), PEPH1(1),
     2 THTA(1), XW(1), YW(1)
      N=NPOINT
      IDM=MXNPT
      INT=10
      CMAX=1.E-12
      SUT 2= SURT(2.)
      PI=3.141592653589793
      TP=2.*PI
      REN=P1/180.
      ETA=376.727
      CuT=-2. *SuT2/(ETA*(1.,1.))
      ZINPUT=CMPLX(0.,C.)
      DO 20 I=1.N
      IP=I+1
      IF(I.EU.N) IP=1
      Dx=XW(IP)-XW(I)
      DY=YW(IP)-YW(I)
      U(I)=SURT(DX#UX+DY*BY)
```

```
20 CONTINUE
    IF(KWRT1.EQ.O) GO TO 804
      WRITE(6,601)
      WRITE(6.701) N
      WRI TE (6.601)
      WRITE (6.732)
      WRITE (6.703)
      WRITE(6.501)
     UO 804 I=1.N
      UO 801 J=1.NFORT
      ISCRPT=IVGS(J)
      IF(ISCRPT.EU.I) GO TC 802
801
    CONTINUE
      WRITE(6,603) I.XW(I).YW(I).U(I)
      GO TH 803
      WRITE(6,604) I,XW(I),YW(I),D(1),VGS(J)
80.2
      IF(I.LT.N) GG TO 804
803
      WRITE(5.601)
804 CONTINUE
    DO 22 I=1,N
    x(I)=TP*Xw(I)
    Y(I)=TP*Yw(I)
    1) (T) = TP +0(T)
 22 CONTINUE
    00 25 I=1.NPORT
    DC 25 J=1.NPURT
    YMMHD(I.J) = (0..0.)
 25 CONTINUE
    UO 100 I=1.N
    DO 100 J=1.N
    C(I.J)=(.0..0)
100 CONTINUE
    DO 150 I=1.N
    CALL SMM1(D(I).P11.P12)
    C(I \bullet I) = C(I \bullet I) + PII
    j=I+1
    TE(I_EW_N)J=1
    C{J,J}=C{J,J}+P11
    C(1, J)=C(1, J)+P12
    C(J,I)=C(J,I)+P12
150 CONTINUE
    UO 200 I=1.N
    ⊢i≃I
    IM = I - 1
    IH(I.Ew.1) IM=N
    [P=I+1
    [F(I.EU.N) [P=1
    (1)u=1u
    UM=D(IM)
    CALL SMM2(X(IM).Y(IM).X(I).Y(I).X(IP).Y(IP).DM.DI.
   2INT, U11, U12, U21, G22)
    C([,I]=C([,I]+421
    C([M, I] = C([M, I]) + C[I]
    C(IM.IP)=C(IM,IP)+612
   C(I, IP) = C(I, IP)+422
    DK= (UM+DI)/2.
    DOD=100.*ABS(DI-EM)/UK
    IF(DUG-LT-3-1G8 TO 160
    CALL ZMM2(X(IP),Y(IP),X(I),Y(I),X(IM),Y(IM),DI,DM,
   2INT, Q11, Q12, Q21, Q22)
160 CONTINUE
    C(I,I) = C(I,I) + Q21
    C(I,IM)=C(I,IM)+G22
    C(IP.I)=C(IP.I)+411
    C(IP, IM)=C(IP, IM)+612
```

```
200 CONTINUE
     IF(NaLTa4)GD TO 200
     N1 = N-1
     N2 = N - 2
     N 4= N-4
     00 250 J=1,N1
     1 = 3 - 1
     IF ( J. Eq. 1 ) I = N
     FI = I
     1) [ = 1) ( [ ] )
     x = x (T)
     Y1=Y(1)
     X2 = X(J)
     Y2=Y(J)
     KA = J + I
     Kn=KA+N4
     09 240 K=KA+KB
     IF(K.GT.N)GO TO 240
     FK±K
     ∂K≃D(K)
     X3=X(K)
     Y3=Y(K)
     L=K+i
     If (K. EQ. N) L=1
                                                                                  0104
     X4=X(1)
                                                                                  0105
     Y4=Y(L)
                                                                                  0106
     CALL SMM3(X1,Y1,X2,Y2,X3,Y3,X4,Y4,DI,DK,Q11,Q12,Q21,Q22)
                                                                                  0107
    C(I,K)=C(I,K)+C11
                                                                                  0108
    C(J.L)=C(J.L)+622
                                                                                  C109
     C(I,L)=C(I,L)+Q12
                                                                                  0110
     C(J,K)=C(J,K)+C21
                                                                                  0111
240 CONTINUE
                                                                                  0112
250 CONTINUE
                                                                                  0113
260 CONTINUE
                                                                                  0114
     KLCRT=6
     IF ((MAUM.Eq. 0) . GR. (NPORT. Eq. 1)) GO TO 268
    NAD=1
     DO 283 JJX=1.NPORT
    KLCRT=KLCRT+1
     ISCRPT=IVGS(JJX)
    DO 265 I=1.N
    CJ(I)=(0.,0.)
265 CONTINUE
    CJ(ISCRPT)=-VGS(JJX)
    GO TO 275
268 NAD=0
    KLCRT=KLCRT+1
    DU 271 I=1.N
    CJ(I)=(0.,0.)
271 CONTINUE
    DO 274 I=I, NPORT
    ISCRPT=[VGS(I)
    CJ(ISCRPT) =-VGS(I)
274 CONTINUE
275 IF(KLCRT.GT.1) GC TO 277
    CALL CROUTI(C,CJ,N,IDM,O,O)
    60 TU 280
277 CALL CRCUT2(C,CJ,N,IDM,0,0)
280 IF(NAD-E4-0) GO TO 286
    UU 283 JJY=1,NPCRT
    JSCRP T= I VGS (JJY)
    YMMHO(JJY,JJX)=1000.*(CJ(JSCRPT)/VGS(1SCRPT))
283 CONTINUE
    GO TO 268
286 IF(NPORT.GT.1) GO TO 289
    YMMHU(1,1)=1000.*(CJ(ISCRPT)/VGS(ISCRPT))
    ZINPUT=CMPLX(1000.+0.)/YMMHO(1.1)
```

```
289 CONTINUE
    KIM=KI-1
    OPH=360 A / ELOAT (KIM)
    AEPHMX=0.
    DO 390 NPH=1.KI
                                                                                0129
    FPH=NPH-1
                                                                                0130
    PHS= JPH*FPH
    THTA(NPH)=PHS
                                                                                0131
    PHR= 0174533*PHS
                                                                                0132
    CPH=COS(PHR)
                                                                                0133
    SPH=SIN(PHR)
                                                                                0134
    DO 300 1=1.N
                                                                                0135
    EPP(I)=(.0..0)
                                                                                0136
300 CONTINUE
    DO 350 I=1.N
                                                                                3137
                                                                                0138
    X\Delta = X(T)
    YA=Y(T)
    IP=I+1
    If (IntwaN) IP=1
    XB=X(IP)
    YB=Y(IP)
    00=0(I)
    CALL CFF (xA, YA, x8, Y8, CL, CPF, SPH, EP1, EP2)
    EPP([]=EPP([]+EP1
    EPP(TP) = EPP(TP)+EP2
350 CONTINUE
    EPS=(.J..O)
    JU 384 I=1.N
    EPS=EPS+CJ(I) # EPP(I)
384 CONTINUE
    EPAB=CABS(EPS)
    IF(EPAB.GE.AEPHMX) ÄEPHMX=EPAB
    ALPHI (NPH) = EPAH
    REPS=REAL(EPS)
    XEPS=ALMAGLEPS !
    IF(EPAB.LE.CMAX) GC TC 385
    PEPHI (NPH) = ATAN2 (XEPS + REPS // RUN
    GC 70 387
385 PEPH1 (NPH)=0.
387 CONTINUE
    IF(KWRT2.E4.0) GO TO 390
      IF(NPH-GT-11 GO TO 366
      wRITE(5,601)
      WRITE(5.710)
      WRITE(5,601)
      WRITE(6.711)
      ARITE(6,601)
      WRITE(6,505) THTA(NPH), REPS, XEPS, EPAB, PEPH1(NPH)
388
      IF(NPH-LT-KI) GO TO 390
      wRITE ( &. 631)
390 CONTINUE
    LPTEPH=1
    IF (AEPHMA.LE.CMAX) GU TO 425
    UD 424 KL=1.KI
    AEPH2=AEPH1(KL)
    ALPH=AEPH2/AEPHMX
    RATIO(KL)=ALPH
    IF(ALPHALE.CMAX) GO TC 423
    ALPHU8=20. *ALUG10(ALPH)
    IF(ALPHOB.LE.-4C.) GO TO 423
    AEPHI (KL)=ALPHOB
    GO TO 424
423 AEPHI(KL)=-40.
424 CUNTINUE
    GO TO 426
```

```
425 LPTEPH=0
426 CONTINUE
    IF(LPTEPH.EQ.O) GO TO 475
    TECKWRT3.FO.CL GO TO 485
      WRITE(5.601)
      WRITE ( 6.714)
      ARTTE(6,601)
      wRITE(6.715)
      WRITE (5.601)
      JU 430 KL=1,KI
      WRITE (6,608) THTA(KL), AEPHI(KL), RATIO(KL), PEPHI(KL)
      IF(KL.LT.KI) GC TO 430
      #RITE(6.6021
430 CONTINUE
    RETURN
475 WRITE (6.718) AEPHMX
485 CONTINUE
    RETURN
601 FURMAT(/)
602 FORMAT(//)
603 FORMAT(19.3F11.4)
604 FURMAT(19.3F11.4.F20.4.F15.4)
605 FURMAT(F14.1,3E15.6,F15.2)
608 FURMAT(F14.1,2F15.6,F15.2)
701 FORMAT(5x,41HNUMBER OF POINTS DESCRIBING THE CYCINDER=13)
702 FURMAT(5x,37HGECMETRY OF CYLINDRICAL CROSS SECTION,9X,22HDRIVING P
   ICINT VOLTAGES)
                                                                   RECVI
                                      Y. WVL.
                                                 D. WVL..SX.30H
703 FURMAT(4x,38HPDINT
                           A. nvL.
   1, VOLTS IM(V), VOLTS)
710 FORMATISX.29FRADIATION PATTERN (RELATIVE))
711 FORMAT(5%, 9HPHI, DEG.,6%,7HRE(EPH),8%,7HIM(EPH),7%,8HMAG(EPH),4%,
   1 11HPHASE, DEG. 1
714 FORMAT(5%, 25HRADIATION PATTERN (DB AND E/EMAX))
715 FORMAT(5x,9HPHI, DEG.,7x,8FEPH, DB.,9X,6HE/EMAX,4X,
   2 11 HPHASE, DEG.)
7.18 FORMAT(5%,7HAEPHMX=E15.8,34H, THUS, NO CB VALUES WERE COMPUTED)
    FND
```

Subroutine SPLFIT

```
SUBROUTINE SPLEIT(MAXP, MAXCO, NPTIN, ISTAPT, PNTIN, IDIVD,
 1 ELEN, COEF, SPNTIN, XPNT, YPNT, APCINT, XW, YW)
   DIMENSION PHTIN (MAXP, 21, SPNTIN (MAXP, 2), COEF (MAXCO, 4, 2),
  1 ELEN(1). [U[VU(1). XPNT(1). YPNT(1). XW(1). YW(1).
  NSEG=NPTIN-1
   DO 10 I=1. MPTIN
   K=(ISTART-1)+I
   IF(K.LE.NPTIN) GO TO 9
   K=(ISTART-NPTIN)+I
 9 SPNTIN(I,1)=PNTIN(K,1)
   SPNTIN(I,2)=PNTIN(K,2)
10 CONTINUE
   CALL SPFIT2 (MAXP, MAXCC, NPTIN, SPNTIN, COEF, ELEN)
   I V=1
   DO 17 I=1.NSEG
   IVK = \{ISTART-1\}+I
   IF(IVK.LE.NSEG) GO TO 14
   IVK=(ISTART-NSEG)+I-1
14 ICV=IDIVD(IVK)
   00 17 J=1, IDV
   JM1 = J - 1
   T=(FLOAT(JM1)/FLOAT(IDV))*ELEN(I)
   T2=T*T
```

```
T3=T*T2
   XPNT(IV)=T3*COEF(I.1.1)+T2*CCEF(I.2.1)+T*CGEF(I.3.1)+COEF(I.4.1)
   YPNT([V]=T3*CDEF([,1,2]+T2*CCEF([,2,2]+T*COEF([,3,2]+COEF([,4,2]
   IV=IV+1
17 CONTINUE
   NPO INT = IV-1
   LXX=ISTART-!
   LXSUM=0
   00 19 I=1.LXX
   LXSUM=LXSUM+IDIVC(I)
19 CONTINUE
   DO 22 I=1. NPGINT
   J = LXSUM + T
   IF(J_GT_NPOINT) J=J-NPCINT
   (I)TM9X=(L)WX
   Yw(J) = YPNT(T)
22 CONTINUE
   RETURN
   ENO
```

Subroutine SPFIT2

```
SUBROUTINE SPFITZ(MAAP, MAXCC, N. PNT, COEF, ELEN)
C
C
       THIS SUBROUTINE COMPUTES THE PARAMETRIC CUBIC SPLINE COEFFICIENTS
C
       TO APPROXIMATE A SMOOTH CURVE THROUGH A 2D SET OF POINTS PNT(1,1)
C
       AND PNT(1,2).
                          THE ARC LENGTH ELEN(I) IS APPROXIMATED TO BE
C
      THE EUCLIDEAN DISTANCE BETWEEN CONSECUTIVE POINTS
C
      MAXP=MAX. NUMBER OF INPUT PUINTS ALLOWED
C
C
      MAXCU=MAX. NUMBER OF SPLINES ALLOWED
C
      NEACTUAL NUMBER OF INPUT PCINTS
€
¢
      EXAMPLE-SUPPOSE 20 PUINTS TO BE INPUT, CALLING PROGRAM HAS FOLLOWING
¢
      DIMENSION PNT (50,2), CUEF (50,4,2), ELEN(5C)
      THEN N=20; MAXP=50; MAXCC=50
C
      DIMENSION PNT(MAXP,2), CUEF (MAXCU,4,2), ELEN(1)
      NUNE=1
      N1=N-NONE
      I nO = 2 .
      0!4E=1.a
      THREE=3.
Ċ
      UO 10 1=1.N1
      IP=I+NONE
      X1=PNT(IP,1)+PNT(I,1)
      Y1=PNT(IP,2)-PNT(I.2)
      EL1=x1*x1+y1*y1
      ELEN(I) = SURT(EL1)
   10 CUNTINUE
      FORM A-MATRIX, SUPER DIAG. ELEMENTS ARE IN COEF(1,4,1), SUB DIAG.
С
      ELEMENTS IN COEF(1,2,1), DIAG. ELEMENTS IN COEF(1,1,1). ELEMENTS
C
C
      CF COLUMN VECTOR FOR X EQUATION IN COEF(1,3,1) AND Y EQUATION IN
(
      CCEF(1,3,2)
      COEF(1,2,1)=0.
      CDEF(1,1,1)=TwG
      CLEF(1,4,1)=ONE
      CUEF(N, 2,1)=CNE
      COEF(N.4.1)=0.
      COEF(N.1.1)=TWO
```

```
X1=THREE/ELEN(1)
      Y1=THREE/ELEN(N1)
      COEF(1.3.1) = X1 * (FNT(2.1) - PNT(1.1))
      CUEF(1,3,2)=X1*(FNT(2,2)-PNT(1,2))
      COEF(N,3,1)=Y1*(FNT(N,1)-PNT(N1,1))
      CUEH(N, 3,2)=Y1*(FNT(N,2)-PNT(N1,2))
      00 20 I=2.N1
      IM= I-NUNE
      IP=I+NONE
      CUEF(I, 2, 11=ELEN(I)
      COEF([,1,1)=TWO*(ELEN(I)+ELEN(IM))
      CUEF(I,4,1)=ELEN(IM)
      x1=THREE/(ELEN(I)*ELEN(IM))
      X2=ELEN(IM)*ELEN(IM)
      X3=ELEN([] *ELEN(])
      COEF(1,5,1)=X1+(X2*(PNT([P,1)-PNT([,1)]+X3+(PNT([,1)-PNT([M,1])))
      COEF(I,3,2)=X1*(X2*(PNT(IP,2)-PNT(I,2))+X3*(PNT(I,2)-PNT(IM,2)))
   20 CUNTINUE
С
      THE A-MATRIX IS TRIDIAGONAL, THE A-MATRIX WILL BE DECOMPOSED INTO
C
      AN UPPER LIAG. MATRIA U AND A LOWER DIAG. MATRIX L SUCH THAT LU=A
¢
      THE DECOMPOSITION OF A INTO L AND U DOES NOT DEPEND ON B
C
      A VECTOR Z FOR EACH X AND Y EQUATION WILL BE COMPLTED SUCH THAT
C
               L FOR X IN COEF(I, 3,1) AND FOR Y IN COEF(I, 3,2)
      1/= 0.
Ç
      COEF(1,3,1)=CUEF(1,3,1)/COEF(1,1,1)
      CUEF(1,3,2)=COEF(1,3,2)/CUEF(1,1,1)
      COEF(1,1,2)=COEF(1,1,1)
      U0 50 I=2.N
      IM= I-NUNE
      COE+(1,2,2)=COEF(I,2,1)
      CBEF(IM, 4, 2) = CDEF(IM, 4, 1)/CDEF(IM, 1, 2)
      COEF(I,1,2;=CDEF(I,1,1)-CUEF(IM,4,2)*CDEF(I,2,2)
      CCEF([,3,1)=(CCEF([,5,1)-CCEF([,2,2)*CCEF([M,3,1))/CCEF([,1,2)
      CCEF(1,5,2)=(CGEF(1,3,2)-CCEF(1,2,2)*CCEF(IM,3,2))/COEF(1,1,2)
   SC CONTINUE
C.
                                                            R IN COEF (1.3.1)
      COMPUTE VECTOR R BY BACK SUBSTITUTION WHERE UR=Z
                                            THE R VECTOR IS THE FINAL
      FOR X AND IN COEF(1,3,2) FER Y
Ć
      SOLUTION VECTOR FOR THE SLUPES
Ç
      DU 60 I=1.N1
      NN=N-I
      CGEF(NN,3,1)=COEF(NN,3,1)-CUEF(NN,4,2)*CDEF(NN+1,3,1)
      COEF(NN, 3, 2) = CCEF(NN, 3, 2) - CCEF(NN, 4, 2) * CCEF(NN+1, 3, 2)
   SO CONTINUE
C
                                   FOR I-TH SEGMENT- X EQ. IS
      COMPUTE CUBIC COEFFICIENTS
Ĉ
      X(T)=CUEF(I,1,1)*T****+CUEF(1,2,1)*T**2+COEF(I,3,1)*T+COEF(I,4,1)
C
      DU 70 I=1,N1
      EL1=JNE/ELEN(I)
      EL2=ELi*EL1
      EL3=cL2*EL1
      IP=I+NONE
      X1=PNT(I,1)-PNT(IP.1)
      Y1=PNT(I,2)-PNT(IP,2)
      COEF(1,1,1)=TwC*EL3*X1+EL2*(COEF(1,3,1)+COEF(IP,3,1))
      COEF(I,1,2)=Twu#EL3*Y1+dL2*(COEF(1,3,2)+COEF(IP,3,2))
       CUEF(I,2,1) = THREE * ELZ*(-X1) - ELI*(TWO*COEF(I,3,1) + COEF(IP,3,1))
       COEF(I, 2, 2)=THREE*EL2*(-Y1)-EL1*(TWO*COEF(I, 3, 2)+COEF(IP, 3, 2))
      CGEF(I,4,1)=PNT(I,1)
      COEF(I,4,2)=PNT(I,2)
   70 CONTINUE
       RETURN
       ENU
```

Subroutine CVRTV

SUBROUTINE CVRTV(APTIN, NPCRT, JVGS, IDIVC, IVGS) DIMENSION JVGS(1), IVGS(1), ILIVU(1) NSEG=NPTIN-1 ISUM=1DO 8 I=1.NSEG DO 6 J=1.NPORT JVG=JVGS(J) IH(JVG.EQ.I) GO TO 4 60 TO 6 4 IVGS(J)=ISUM J=NPORT 6 CONTINUE ISUM=ISUM+IDIVC(I) 8 CONTINUE RETURN END

Subroutine CROUT1

```
SUBROUTINE CROUTICE, S, N, ILM, ISYM, IWR)
      SET ISYM = O IF MATRIX IS SYMMETRIC
C
      SET IWE = 0 OR NEGATIVE TO AVOID WRITECUT
C
      COMPLEX C(IDM, IDM), S(IDM)
      CUMPLEX SS
      COMPLEX F.G.H.P.T.U
      FORMAT(1x,1110,1615,3,1615,6,2615,4)
      FUF MAT(1H0)
      IF(N.EQ.1)S(1)=S(1)/C(1.1)
      I+(N.E4.1)60 TC 100
      S OF DOLC . SMY SI HI
      DO 6 I=1.N
      U0 6 J=I.N
      C(J,I)=C(I,J)
    CONTINUE
    CONTINUE
      F=C(1,1)
      UU 13 L=2,N
      G=C(1,L)
 10 C(1+L)=G/H
     00 20 L=2.N
     LLL =L-1
      DU 20 I=L, N
     f=((1,L)
     00 11 K=1, LLE
     G=C(1.K)
     H=C(K,L)
 11 F=F-6*H
     C(I_1L)=F
     If-(L.EW.I) 60 TG 20
     P=C(L,L)
     IF(ISYM.EW.O)GC TO 15
     F=C(L,I)
     UN 12 K=1,LLL
     G=C(L,K)
     H=C(K+I)
```

```
12 F=F=6*H
    C(L, I)=F/P
    GO TO 20
15 F=C(I,L)
    C(L.1)=F/P
 20 CONTINUE
    60 30 L=1.N
    P=C(L+L)
    T=S(L)
    IF(L.Ev.1) GO 70 20
    LLL=L-1
    ยถ 25 K=1•LLL
    1- =C(L +K)
    U=S(K)
25 T=T-F*J
30 S(L)=T/P
    03 38 L=2•N
    1 = N - L + 1
    II = I + 1
    T=S(I)
    00 35 K=II.N
    F=C(I,K)
    H=STK L
35 T=T-r+u
38 S(1)=T
    IFILIME. LE. 3) GC TO 53
    CNOR=.0
    03 43 I=1.N
    55=5111
    SA=CABSISSI
     IF (SA.GT.CNOP) CNCR=SA
40 CONTINUE
    DU 44 I=1.N
     $5=$(I)
     SA=CABS(SS)
     SNOR=.0
     IF(CNUR.GT.O.)SNUR=SA/CNOR
     SR=REAL (SS)
     SI=AIMAG(SS)
     C•=H9
     IF(SA.UT.O.)PH=57.29576#ATANZ(SI.SR)
     WRITE (6.2) I. SNUR. PH. SR. SI
44 CONTINUE
     wPITE(6,5)
50 CONTINUE
 100 GONTINUE
     RETURN
     ENU
```

Subroutine CROUT2

```
SUBROUTINE CROUT2(C,S,N,IDM,ISYM,IBR)

SET ISYM = 0 IF MATRIA IS SYMMETRIC

SET IWR=0 OR NEGATIVE TO AVOID WRITEOUT COMPLEA C(IDM,IDM),S(IDM)

COMPLEA SS

COMPLEX F,G,H,P,T,U

2 FORMAT(IX,1110,1115,2,1115,0,2115,4)
```

```
FORMAT(1HO)
    DC 30 L=1.N
    P=C(L.L)
    T=S(L)
    IFILABUALIGO TO 35
    LLL=L-I
    00 25 K≃1+LLL
    F=C(L+K)
    U=${K}
25 T=T-F*U
30 S(L)=T/P
    0u 38 L=2∙N
    I=N-L+i
    11=1+1
    T=3(T)
    00 35 K=II•N
    F=C(I+K)
    U=S(K)
   T=T-F≭U
3.5
38 S(I)=T
    IF(INFALE O) GC TO 50
    CNOR=.J
    01 43 I=1.N
    SS = SIII
    SA=CABS(SS)
    IF ( SA. GT. CHOR ) CNCR=SA
40 CONTINUE
    0.1 44 I=1.N
    5S=S(1)
    SA=CABS(SS)
    รัพประเอ
    IF(CNOR.GT.J.) SNOR=SA/CNOR
    SR=REALISS /
    SI=AIMAG(SS)
    PH= .0
    IF(SA.CT.O.)PH=57.29578*ATAN2(SI,SR)
    WRITE(6,2) I, SNCR, PH, SR, SI
44 CUNTINUE
    wRITE (0.5)
50 CONTINUE
100 CONTINUE
    RETURN
    END
```

Subroutine SMM1

```
SUBROUTINE SMM1(DK,ZM11,ZM12)
CCMPLEX ZM11,ZM12
CCMPLEX H0,H1
DATA PI/3.14199/
CDK=COS(DK)
SDK=S[N(DK)
CALL HANK(DK,HC,H1)
SDKS=SUK**2
CUKS=CUK**2
ZM11=2.**H1*CUK-FC*SUK-2.**(.0,1.)*(1.+CUKS)/PI/DK
ZM12=H3*CUK*SUK-H1*(1.+CUKS)+4.**(.0,1.)*CDK/PI/DK
ZM12=H5.**UK*ZM11/SUKS
ZM12=15.**UK*ZM12/SUKS
RETURN
END
```

Subroutine SMM2

```
SUBROUTINE SHM2(X1, Y1, X2, Y2, X3, Y3, DK1, UK2, INT, S11, S12, S21, S22)
COMPLEX SX1, SX2
COMPLEX $11.812.821.822
CGMPLEX G11.G12.G21.U22
 COMPLEX HOSHISHECSHHI
 COMPLEX F(3)
 COMPLEX AA(11,51,8A(11,5),CA(11,3),ZA(11,13)
CCMPLEX Ab(11,5),68(11,5),C8(11,3),Z8(11,13)
COMPLEX AC(11.5).BC(11.5).CC(11.3).ZC(11.13)
 CCMPLEX AU(11,5),BU(11,5),CU(11,3),2U(11,13)
 EQUIVALENCE (ZA(1,1),AA(1,1),(ZA(1,2),BA(1,1)),(ZA(1,11),CA(1,1))
 EQUIVALENCE (ZE(1,1), Ab(1,1)), (ZB(1,6), BE(1,1)), (ZB(1,11), CB(1,1))
 EQUIVALENCE (20(1,1),40(1,4)),(20(1,6),80(1,1)),(20(1,11),00(1,1))
 EWUIVALENCE (ZD(1,1),AD(1,1)),(ZD(1,6),BD(1,1)),(ZD(1,11),CD(1,1))
 DATA P1/3-14159/
 DATA 6.H/.314155..261759/
 NAA ATAG
                                             1.( 0.0
                                                          , 0.0
. ( J.)
           . 0.0
                      1.1 C.C
                                  . 0.0
                                                                    1.
                      1,60.0
                                                          , 2.0
           0.0
                                  , 0.0
                                             1.1 3.0
                                                                    1.
. ( 0.5
.1 0.0
           , 0.0
                                  + C.C
                                                          . 0.0
                      Jet Jet
                                             1.1 0.0
                                                                    1.
                                                          . 0.0
           0.0
                      1.1 0.0
                                  . 0.0
                                             1, ( 0.0
                                                                    1 .
. ( 0.0
-(-3.J00157,-0.005209);(-6.C01205,-C.013471);(-0.J03419,-0.014947);
• (-J.035387,-0.017790), (-J.009293,-0.018725), (-J.011502,-D.018116),
• (-0.012932,-0.016578), (-0.013971,-0.014465), (-0.015021,-0.011636).
. (-). )1c 041.-0.007€721.€ 0.0
                                  • 0•0
                                             1.(-0.000625,-0.0202001.
. (-J.JJ-508,-0.040586), (-G.G12827,-0.057989), (-0.024041,-J.D69192),
.(-J.J35087,-J.C73050),(-0.043494,-C.07C898),(-0.048825,-C.C65C75).
. (-J. U52534, -C. C57103), (-O. C56248, -C. 046633), (-O. C60114, -O. 032123).
                      1. (-J. CC1251. -C. 043121).(-0. CC9051.-0. G86548).
al Cat
           . 0.0
• (-0.025867,-C.123838), (-C.C.C48756,-O.149317), (-O.071481,-O.157381),
.(-J. 088823,-0.153436),(-C. 099479,-C. 141431;(-C. 16159,-0.125014),
.(-J.112630,-C.104181),(-C.120115,-J.076060),( J.O.
                                                         , 0.0
a(-0.001879, -0.071060), (-0.013644, -0.142381), (-0.039218, -0.204035),
• (-J• 074406,-C• 245482), (-O• 105840,-C• 262081), (-O• 136913,-O• 256958),
. (-0.152731,-0.237878), (-0.160875,-0.211618), (-0.167710,-0.179818).
· (-3.170967,-0.1350491/
DATA BA7
. ( 0.0
                      ), (-0. G02341, -C. 100292), (-0. C17072, -0. 200445),
           . 0.0
• (-0.049401,-0.287561), (-0.094491,-0.347606), (-0.140551,-0.375563),
•(-0.175815,-0.368409),(-0.19_020,-0.342136),(-0.201470,-0.305169),
• 1-0.203665, -0.2624761, 1-0.208911, -0.2120541, t 0.0
                                                          , 0.0
\bullet (-0.002513, -0.126603), (-0.018412, -0.252138), (-0.053664, -0.361786),
-(-3.103547, -0.439113), (-0.155290, -0.474822), (-0.194970, -0.470681),
•(-0.214671,-C.437498),(-0.215903,-0.386772),(-0.207611,-0.333925),
                                 . 0.0
• (-0.200231,-0.274917), ( G.G
                                            1,(-0.002348,-0.145669),
• (-3.017286,-6.288743), (-C. C50747,-0.413776), (-0.098784,-0.503510),
•(-J•149377,-C•547C42),(-J•180217,-O•5441541,(-O•205396,-O•504588),
a(-J.199632,-0.443077),(-C.177890,-0.372977),(-0.151496,-0.301724);
a ( 9.0)
           . 0.0
                      1,1-0.001890,-0.153463),1-0.013976,-0.302360),
• (-0.0%1308,-0.431869), (-U.CB1075,-0.525(34), (-0.123553,-0.572360),
• (-0.156179,-C.569787), (-0.168778,-C.525037), (-0.157527,-0.451540),
• (-0.120031,-0.364194), (-0.083294,-0.274971), ( 0.0
                                                        . 0.0
·(-0.001252,-0.146683),(-0.005364,-0.286862),(-0.027837,-0.407352),
•(-0.J55025,-0.494303),(-u.C84419,-0.537583),(-0.106997,-0.533585),
•(-J•114525,-0•486425),(-J•102426,-O•406260),(-O•071196,-O•306432),
. (-0.025934,-0.200279)/
 LATA CA/
. ( 0.0
           . C.C
                      1, (-0.300631,-0.1231931,(-0.004700,-0.238814),
• (-0.014034,-0.356302), (-0.C278>3,-0.405296), (-0.043016,-0.438105),
•(-0.054629,-0.431450),(-0.657994,-0.387153),(-0.049893,-0.311671),
```

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•(-0.029555.+C.214744).( 0.C01190.-C.107643).( C.O
                                                         . 0.0
• (-0.000169,-0.002703), (-0.001262,-0.158761), (-0.003760,-0.221141),
a(-0a007539,-0a262649), (-CaC11564,-0a281895), (-0a014831,-0a273976),
.(-0.015658,-0.240766),(-0.013104,-0.185809),(-0.006753,-0.114839).
                                 , C.O
.( 0.003082,-0.035048).( 0.0
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. ( 0.0
           , 0.G
.(-J.)00330,-C.004419),(-J.002298,-C.009053),(-D.006166,-O.012438),
.(-J.J10691,-0.013477),(-0.014254,-0.012457),(-0.016104,-0.010646),
.(-0.016618,-0.009173),(-0.016583,-C.CO8348),(-0.01641C.-0.007906).
                                            1.(-0.001233.-0.017091).
.(-J.J16041,-0.007672),( J.C , 0.0
.(-0.008606,-0.035027),(-0.023189,-0.048331),(+0.040403,-0.052781),
.(-U.)541CC.-C.C49271),(-C.O61234.-O.O42525),(-O.C63056.-O.O35971).
. (-0.062630,-0.034306), (-0.061551,-0.032676), (-0.060114,-0.032123),
                     ), (-0,002470,-0,036321),(-0,017311,-0,074457),
0.0
           , 0.0
.(-0.046949,-0.103416),(-0.082436,-0.114316),(-0.111144,-0.108359),
• 1-0.126161,-0.094901), 1-0.129404,-C.083489), (-0.127208,-0.077792),
                                                      , 0.0
• (-0.123999,-0.076101), (-0.120115,-0.076060), ( 0.0
.(-0.003713,-0.059481),(-0.026160,-0.121893),(-0.071555,-0.170647),
• (-0.126915,-0.191508), (-0.172655,-0.184997), (-0.196694,-0.164838),
· (-0.200504,-0.146686), (-0.193918,-0.138021), (-0.185403,-0.137040),
.(-0.176967,-C.139049)/
CATA BBZ
                     1, (-0.004629, -C.083280), (-0.032827,-0.170404).
. ( 0.0
.(-0.090683,-0.240484),(-0.162744,-C.274343),(-0.223733,-0.270515),
. (-J. 255911, -O. 245296), (-O. 258573, -C. 220C35), (-O. 243987, -O. 207409),
.(-0.225543,-C.207119),(-C.206911,-C.212054),( 0.0
                                                       , 0.0
.(+J.006975, +C.104115),(-0.C35511, -C.212304),(-C.099148, +D.301627),
• (-0.186198,-0.349537), (-0.253340,-0.351579), (-0.287750,-0.323805),
.(-0.287394,-C.291119),(-0.261916,-C.271716),(-0.229114,-0.268977),
.(-0.200230,-0.2749171,( J.J. , 0.0
                                            ),(-0.004651,-0.116450),
· (-J. 955441, -0. 240050), (-U. 0943c), -C. 342317), (-C. 173687, -0. 401796),
. (-0.244239, -0.411004), (-0.20159), -0.382965}, (-0.277550, -0.342643),
.(-0.241536,-0.312267),(-0.193866,-C.300520),(-0.151496,-0.301724),
                     1,1-3.005747,-C.123174),(-0.027115,-0.247208),
           . 0.0
.( 0.0
.(-0.077274,-0.352148),(-0.143916,-0.41642C),(-0.204527,-0.430762),
.(-0.236603,-0.4C3660),(-0.229491,-C.356579),(-0.189144,-0.512600),
.(-0.133703,-0.285447),(-C.C83294,-C.274971),( 0.0
                                                      , 0.0
                                                                  ),
• (-0.002503,-0.115861), (-0.C18211,-0.229262), (-3.052342,-0.324C15),
. (-0.098474,-0.382853), (-0.141226,-C.397007), (-0.163819,-0.370954),
• (-0.150510,-0.32C632), (-C.121514,-0.266362), (-0.C72CC4,-0.224233),
.(-J.025934,-C.20G2791/
 DATA CB/
                     ),(-0.001253,-0.094807),(-0.009157,-0.183962).
ت•ن ا•
           , 0.0
. (-J.026492,-J.255549), (-C.U5C250,-O.298C99), (-O.O72543,-9.305734),
.(-0.084314,-0.281122),(-C.C79517,-C.234818),(-0.058378,-0.181876),
.(-0.027802,-C.136541),( C.001190,-O.107643),( O.0
                                                        . 0.0
.(-J.J00336,-C.C58491),(-O.GJ2462,-O.110535),(-C.007153,-O.148969),
• (-)•013629;-0•168440); (-0•019770;-0•167176); (-0•023005;-0•147751);
.(-0.021507,-0.11c582),(-0.01c160,-0.082239),(-0.005851,-0.053094),
                                                         . 0.0
. ( . J. 203082, -0.0350471, ( 3.0
                                  . 0.0
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.1 0.000107, 0.005062),1 0.001201, 0.005480),1 0.003357, 0.011733),
.( J.005987, 0.010617), ( 0.007728, 0.005929), ( 0.007122, -0.001153).
.( 0.003386,-0.CC8306),(-0.C03018,-0.012842),(-0.010334,-0.012768),
                                           ),( 0.000625, 0.019708).
•(-0.016041,-0.007672),( 0.0 , C.0
.( 0.004494, G.C36757), ( 0.C12606, O.C456C6), ( 0.022612, 0.041617),
.( ).029476, 0.0239481, ( 0.027707, -0.0030901, ( 0.014200, -0.030841),
• (-0.009630,-0.049148),(-0.037502,-0.050247),(-0.060114,-0.032123),
        , 0.0 1,1 U.C01251, C.0420771,1 0.009025, 0.0784341,
. ( 0.0
.( 0.025459, C.C97667),( C.C46091, C.090318),( 0.061025, 0.054388).
.( 0.059152,-0.001720),( 0.033682,-0.060742),(-0.013504,-0.101986),
.(-0.070820,-0.108932),1-C.120115,-C.076060),( 0.0
                                                      . 0.0
. ( 0.001878, 0.069353), ( U. C13609, 0.129150), ( 0.038676, 0.161592),
.( U.07087C, 0.151890),( C.C95735, 0.096691),( 0.096450, 0.008192),
• ( 0.061778, -0.087762), (-J.007530, -0.159430), (-0.095565, -0.180229).
•1-0-176966,-C-1390491/
 CATA BC/
                     ),( U.CC2340, C.C97909),( 0.017036, 0.182031),
. ( 0.0
           • 0 C
a( 0.048831, 0.228837), ( c.090759, 0.219080), ( 0.125444, 0.148077),
.( 0.131951, C.C3C721),( 0.C94932,-0.1C3692),( 0.012912,-0.205636),
.(-0.098307,-6.248638),(-c.268911,-c.212054),( C.0 , 0.0 ),
.( 0.002512, 0.123636),( C.Cle301, C.2293C4),( 0.053174, 0.289463),
.( 0.100303, 0.282269),( 0.142107, C.202389),( C.156246, 0.066072),
. ( J.125019, -0.091267), ( J.043959, -0.224663), (-0.073914, -J.293479),
.1-0.200230,-0.2749171,1 C.C , 0.0 ),1 0.002348, 0.142316),
.( 0.017264, 0.253045),( 3.050406, 0.332991),( 0.096514, 0.330174),
.( 0.140050, C.249802),( 0.160697, 0.108489),( C.14098C,-0.058585),
. ( 0.074360,-0.207009),(-0.030665,-0.296164),(-0.151496,-0.301725).
.( 0.0 , 0.0 ), ( 0.001890, 0.150007), ( 0.013964, 0.275992), ( 0.041123, 0.349639), ( 0.07904), 0.351277), ( 0.118501, 0.277962),
   0.141190, 0.146142), ( 0.133630,-0.011777), ( 0.089045,-0.156080),
. (
• ( J. J11706, -0.250991), (-0.683294, -0.274971), ( 0.0 , 0.0 ),
.( 0.001252, 0.143458),( 0.009359, 0.262368),( 0.027766, 0.331649),
• ( 0.101866, 0.031562), ( 0.078589, -0.089743), ( 0.033201, -0.172247),
· (-0.025934,-0.200279)/
 DATA CC/
.( 0.0 , 0.0 ),( 0.000631, 0.120521),( 0.004699, 0.218624),
.( 0.014019, 0.274433),( 0.027797, 0.277569),( 0.042655, 0.230437),
.( J.053695, G.147572); ( G.C56225, G.O51447); ( O.C47521,-0.033974);
                                                       , 0.0
• ( 0.027976, -0.069911), ( 0.001190, -0.107643), ( 0.0
. ( 0.000169, 0.086793), ( J. Cúlzbz, 0.144415), ( 0.003779, 0.177640),
.( 0.007537, 0.175275),( 0.011670, 0.141517),( 0.014905, 0.088406),
.( 0.016000, 0.031965),( 0.014226,-0.012695),( 0.009626,-0.035735),
• ( J.003082,-C.035048), ( 0.0
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• ( 0.000330, 0.004351), ( 0.002288, 0.008488), ( 0.005025, 0.010438),

. ( ). (009851 + 0. (08699) + ( 0.011267 + 0.003234) + ( 0.008556 + −0.003546) +

    ( ).002400, -0.0089331, (-0.005336, +0.011122), (-0.011981, -0.010193),
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, 0.0 1,( 0.001233, 0.016831),
  .(-3.016041,-3.007672),( J.O
  .( 0.008573, 0.032883),( 0.022693, 0.040770),( 0.037426, 0.034397),
  .( 0.043448, C.C14346),( U.C34456,-0.011650),( 0.011460,-0.033113),
  .(-0.017799,-0.042916),(-C.C43658,-C.04C804),(-0.060114,-0.032123),
           , 0.0 ). ( 0.002470, C.035786), ( 0.017250, G.070055),
  .( ).0
  .( 0.046051, 0.0879721,1 J.C77302, 0.0768561,1 0.091497, 0.036918),
  . ( 0.076138,-0.017229), ( 0.031338,-0.064626), (-0.027930,-0.089832),
  • (-0.082918,-0.090514), (-0.120115,-0.075050), ( 0.0 , 0.0 ),
  .( 3.003712, 0.058644), 1 0.026082, 0.115048), ( 0.070396, 0.146811),
  . ( 0.119825, 0.133934), ( C.146662, C.074822), ( C.129376, -0.C10454),
  • ( 0.066115, -0.096716), (-0.024054, -0.140662), (-0.112498, -0.153109),
  a(-Ja1769c6,-C.139049)/
   CATA BU/
                      1. ( 0.004628, 0.082184), ( 0.032749, 0.161505),
             . 0.0
  . ( ).)
  .( 0.089520, 0.209794),( J.155640, C.200706),( 0.195878, 0.1293d1),
  .( 0.184909, 0.017794), ( C.113300, -0.096151), ( 0.001649, -0.178288),
  .(-J.115858,-0.214174),(-0.208911,-0.212054),( 0.0 , 0.0 ),
  .( 3.004972, 0.1028731,( 0.035451, C.2022911,( 0.098233, 0.267494),
  .( 0.174443, C.268419),( 0.228664, C.196367),( 0.228486, 0.071031).
  •( 0.162555,-C.C68660),( 0.04c529,-C.1836C8),(-0.085834,-0.251922),
  •1-0.200230,-0.2749171,( 0.C , 0.0 ),( 0.004650, 0.117229),
  .( J. 033406, C. 230289), ( U. 09385J, 0.309461), ( 0.170260, 0.324651),
  .( 0.230890, 0.264202),( 0.244309, C.142450),( 0.195843,-0.006541).
  • ( J.094614,-0.144625), (-G.C31874,-C.245375), (-C.151496,-0.301724),
           , U.O 1,1 U.GOS747, C.1221691,1 3.C271C3, 0.2392441,
  . ( ).)
  . ( 0.077090, 0.3256691, ( 0.142097, C.3550501, ( 0.199560, 0.314887),
  . ( D. 221934, U. 213275), ( U. 195194, U. 075462), ( C. 122278, -0. 066987),
  • ( 0.021307,-0.188174), (-0.083294,-0.274971), ( 0.0 , 0.0
  of 0.002503, 0.115253), (0.018213, 0.224467), (0.052364, 0.308184),
  .t J. J90584, 0.346406), t 0.141473, C.328253), t C.163807, 0.256767),
  .( J.154392, O.147472),( O.112575, C.0221F1),( O.047622,-O.098570),
  . (-0.025933,-C.200279)/
   DATA CD/
                      ), ( 0.001253, 0.094692), ( 0.009162, 0.183015),
  . ( 0.0
             . 0.0
  •( 0.026560, 0.252210),( 0.050653, 0.289695),( 0.074115, 0.287980).
  . ( 0.088393, 0.247268), ( 0.087403, 0.175067), ( 0.069807, 0.083426),
  • ( 0.038988,-0.014601), ( 0.001190,-0.107643), ( 0.0 , 0.0 ),
  .( 0.003336, C.058756),( 0.302464, 0.112516),( 0.007181, 0.154899),
  -4 0-J13806, 0-18C161), ( L-02U446, C-18472C), ( C-C24829, J-1680C7),
  . ( 0.025253, 0.132455), ( C.C21189, 0.082759), ( 3.013315, 0.024850),
  .( 0.003082,-0.035047),( 0.L , 0.9
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                       1.1 0.0
  e t Deu
   CKM=DK1
   IF (DK 2 GT = DK1) DKM=DK2
   IF ( DKM.LT. 3. ) GC TO 10
   Sil=(.0..0)
   $12=(.0,.0)
   521=(.0,.0)
   522=(.0..0)
   RETURN
10 CONTINUE
   DK=(DK1+DK2)/2.
   PD=100 *ABS(DK2+EK1)/DK
   IF(PU.LT.3.)GU TO 42
   CALL ZMM2(X1,Y1,X2,Y2,X3,Y3,LK1,DK2,INT,S11,S12,S21,S22)
   RETURN
42 CONTINUE
   DKS=UK1*UK2
```

```
CDK = CDS CDK I
   SDK=SIN(DK)
   SDK S= SDK ##2
   CCKS=CUK**2
   LBET=(\lambda 2-\lambda 1)/\partial KI
   SBET= (Y2-Y11/UK1
   XB=(X3-X1)*CBET+(Y3-Y1)*SBET
   YB=-(A0-X1) #SBET+(Y3-Y1) #CBET
   CAL = (AB - DK1)/UK2
   SAL=ABSIYBI/UKZ
   CALL HANK(UK.HO.H1)
   IF(CAL.LT. J. ) GG TO 50
   IFISAL.GT.. 341GC TU 50
   CNT=15.*UK*CAL/SCKS
   CALL HANK(2. +DK. FHO. HH1)
    S11=CNT+(2.+HH1+CDK-H0+SDK+CDK-H1+(1.+CDKS)+2.*(.0.1.)*CDK/PI/DK)
    S12=CNT*{2.*HHC*SDK*CDK-2.*HH1*COKS
  2+2. *H1*COK-H9*SCK-2.*(.C.1.)/PI/DK)
    $21=CNT*(2.*H1*CDK+H0*SUK-2.*HH1-2.*(.C.1.1*CDK$/PI/CK)
    $22=CNT*(H0*COK*SDK-H1*(1.+CUK$)-2.*HH0*SDK
   2+2。*Hn1+CÚK+2。*(.0.1。)*CÚK/PI/ÚK)
    RETURN
50 CONTINUE
    F(1) = HO
    S11= UK*(-H1+2.*(.0,1.)*CDK/PI/UK) -
    $1/=- UK*(SCK*HC-CUK*H1+2-*(-0.1-1/PI/OK)
    S21= OK*(CUK*H1-2.*(.0.1.) *CLKS/PI/DK)
    522= DK*CUK*(SEK*HO-CUK*H1+2.*(.0.1.)/PI/DK)
    AL=1.570796
    IF(CAL.NE.O.)AL=ATAN2(SAL.CAL)
    AL=ASSIALI
    IF(4L.GT.2.0)GG TD 82
    TK=0K/2.
    DO 80 J=2.3
    TKS=TK**2
    RK=SWKT(DKS+2.*UK*TK*CAL+TKS)
    CALL HANK(RK+HO+F1)
    F(J) = HO
    TK=DK
80 CONTINUE
    SA=1. +SDK/DK+4.*(CUK-1.)/UKS
    SH=8.*(1.-CDK)/UKS-4.*SUK/UK
    SC=3. *SUK/UK-CCK+4.*(COK-1.)/UKS
    $11=$11+CUK *($C*F(1)+$8*F(2)+$A*F(3))
    $12=$12+CDK *($A*F(1)+So*F(2)+SU*F(3))
    S21=S21-4SC*+(1)+SB*+(2)+SA*+(3))
    522=$22-($A*F(1)+$6*F(2)+$C*F(3))
    GO TO 98
82 CONTINUE
    Sx1=(.0,.0)
    SA2=(.U..O)
    INP=2#(INT/2)
    FIT=INP
    I P= INP+1
    DT=DK/FIT
    TK=.U
    $GI = - 1.
    UU 93 1=1, IP
    D=SGI+3.
    1+(1.60.1)0=1.
```

```
IF( I . EU . IP ) D=1 .
    TKS=TK**2
    RK=SURT (DKS+2. *DK*TK*CAL+TKS)
    CALL HANK(RK+HC+H1)
    SI=SINIDK-TK1
    S2=SIN(TK)
    S \times 1 = S \times 1 + S1 \times H0 \times G
    Sx2=SX2+S2*H0*D
    S61=-S61
    TK=TK+DT
90 CONTINUE
    SX1=SX1*UT/3-
    Sx2=Sx2*BT/3
    S11=S11+CDK*SX1
    S12=S12+CDK*SX2
    S21=S21-SX1
    $22=$24-$X2
93 CUNTINUE
    X=JK/G
    I=4+1.5
    IF( I.LT. 2) [ = 2
    If( [.GT.10) [=13
    IM = I + 1
    [P=[+]
    Y = \Delta I / H
    ته1+1 = ل
    IF(J.LT.21J=2
    1F(J.6T.12)J=12
    JM=J-1
    JP=J+1
    f [ = I
    FJ=J
    AI=FI-1.
    YJ=FJ-1a
    P = X - X I
    ₩=Y-YJ
    PT=P/2m
    UT= U/2.
    A=PT*(P-1.)
    B=P T* (P+1.)
    C=uT*(u-1.)
    0=4T*(4+1.)
    E=1.-P**2-4**2
    G11=A*ZA(IM,J)+B*ZA(IP,J)+C*ZA(I,JM)+U*ZA(I,JP)+E*ZA(I,J)
    612=A+ZB(IM,J)+E+ZB(IP,J)+C+ZB(I,JM)+O+ZB(I,JP)+E+ZB(I,J)
    621=A*ZC(IM,J)+b*ZC(IP,J)+C*ZC(I,JM)+D*ZC(I,JP)+E*ZC(I,J)
    G22=A*ZD(1M,J)+B*ZD(IP,J)+C*ZU(1,JM)+O*ZC(I,JP)+E*ZU(1,J)
    CCC=15./SUKS
    S11=CCC*(CAL*S11+G11+DK)
    $12=CCC*(CAL*$12+G12*UK)
    $21=0CC*(CAL*$21+621+0K)
    $22=CCC*{UAL*$22+G22*UK}
    RETURN
    ENU
```

Subroutine ZMM2

```
SUBROUTINE ZMM2(x1, Y1, x2, Y2, X3, Y3, DK1, CK2, INT, S11, S12, S21, S22)
    COMPLEX RKH1
    COMPLEX SX1.5X2
    COMPLEX $11.$12.$21.$22
    CUMPLEX Y11.Y12.Y21.Y22
    COMPLEX EY1, EY2
    COMPLEX HO, HI, FFC, HF1, SHO, SH1
    COMPLEX DHHO. DHF1. DFC. UH1. USHO. USH1
    COMPLEX 0Y11.0Y12.6Y21.0Y22
    COMPLEX CCP
    COMPLEX FUN
    COMPLEX F(3)
    DATA CCP/(.0..63662)/
    CKM=DK1
    IF (DK 2. GT. UK1) DKM=DK2
    IF(DKM.LT.3.)GC TO 10
    S11=(.0..0)
    S12=(.0,.0)
    S21=(.0,.0)
    522=(.0,.0)
    RETURN
10 CUNTINUE
    SUK1=SIN(DK1)
    SUK2=SIN(DK2)
    CDKI=CUS(DKI)
    CUK 2= CO5 (UK 2)
    CBET=(XZ-X1)/DK1
    SBET=(Y2-Y1)/UK1
    XB = \{X3 - X1\} \times CBET + (Y3 - Y1) \times SBET
    YB=-(X3-X1)*SbeT+(Y3-Y1)*CBET
    CAL=(X8-UK1)/UK2
    SAL=ABS (YB)/DK2
    CALL HANK (DK2, HHC, HHI)
    BHH0=DK2*HH0
    DHH1=DK2*HH1
    C1S2=CUK1≠SUK2
    C1C2=CUK1*CUK2
    IF(CAL.LT.O.)GE TO 20
    IF(SAL.GT..O4)GU TC 20
    CNT=15a*CAL/SDKI/SDK2
    CALL HANK(DK1.HO.H1)
    DH0=DK1 *H0
    DHI=DK1*H1
    DKS=DK1+DK2
    CALL HANK ( DKS, SHC, SH1)
    DSHO=DKS+SHO
    LSH1=UKS*SH1
    S11=CNT*(CDK1*DSH1+C1S2*DH0-C1C2*DH1-DHH1+CCP*CDK2)
    $12=CNT*(COK2*OHF1-SUK2*OHF0-CCP+COK1*OF1+C1$2*D$H0-C1C2*D$H1)
    $21=CNT*($0K2*DHO-DSH1+COK2*DH1+CDK1*OHF1-CCP*61C2)
    522=CMT*(C1S2*DHF0-C1C2*DHH1+CCP*CDK1-DH1-SDK2*DSH0+CCK2*DSH1)
    RETURN
20 CONTINUE
    S11=-OHH1+CCP*CDK2
    S12=-SDKZ*DHHO+CUK2*DHH1-CCP
    S21=(DHH1-CCP*CDK2) #CDK1
    $22=(50K2*UHH0-CCK2*DHF1+CCF)*CUK1
    DKS1=DK1**2
    AL=1.570796
    IF(CAL.NE.D. FAL=ATANZ(SAL, CAL)
```

```
\Delta I = \Delta_{1} S S I \Delta I J
    INP=2*(INT/2)
   FIT=INP
   IP=INP+1
    IF(AL-GT-2-01GO TO 82
    TK=.0
    00 80 J=1.3
   TKS=TK**2
    RK=SURT (UKS1+2. *DK1*TK*CAL+TK$)
    CALL HANK(RK+HO+H1)
    F (JJ) = HÚ
   TK=TK+DK2/2.
85 CONTINUE
    0KS2=0K2**2
    SA=1.+SUK2/BKZ+4.*(CUK2-1.1/CKS2
    Sd=3.*(1.-COK21/OKS2-4.#SDK2/OK2
    SC=3.*SUK2/UK2-CUK2+4.*(CUK2-1.)/OKS2
    S11=S11+CUK1*(SC*F(1)+S0*F(2)+SA*F(3))
    $12=$12+$UK1*($A*F(1)+$B*F(2)+$C*F(3))
    S21=S21-(SC*+(1)+S8*F(2)+SA*F(3))
    $22=$22-($4+(1)+$0*F(2)+$C*F(3))
    Gu TO 98
82 CONTINUE
    DT=DK2/FIT
    TK=.0
    Sx1=(.0,.0)
    SX2=1.0+.01
    SGI =-1.
    00 90 I=1.IP
    υ=SGI+3•
    I+(I.E⊎.1)0=1.
    IF(I.E. IP)0=1.
    TKS=TK**2
    RK=SORT(UKS1+2.*DK1*TK*CAL+TKS)
    CALL HANKIRK . HO. HIL
    S1=SIN(DK2-TK)
    S2=SIN(TK)
    SX1 = SX1 + S1 * H0 * D
    SX2=SX2+S2*H0*D
    SGI =- SGI
    TK=TK+DT
90 CONTINUE
    SX1=SX1*DT/3.
    Sx4=Sx2*DT/3.
    $21=$21-$X1
    $22=$22-$x2
   S12=S12+CDK1*SX2
    S11=S11+CUK1*SX1
98 CONTINUE
    JP=IP
    Y11=(a0.0)
    Y12=(.0+.0)
    Y21=1.0,.01
    Y22=(.0..0)
    B= . 0
    IF(AL.LT..05)GC TO 210
    ALT =AL/2.
    CALT=COS(ALT)
    SALT=SIN(ALT)
    RCP=(UK1+UK2)*CALT
```

```
RSP=(BK2-UK1)*SALT
    PHC = ATAN2 (RSP + RCP)
    SGI=-1.
    PH=-ALT
    DPH=AL/FIT
    UC 200 I=1.IP
    D=SGI+3.
    1F( [ . E( . 1 ) U= 1 .
    IF(I.Eq.IPID=1.
    SAP=SIN(ALT+PH)
    SAM=SIN (ALT-PH)
    IF(PHaleaPHC)RMAX=DK1*SAL/SAM
    IF(PH.GT.PHC)RMAX=UK2*SAL/SAP
    URK =RMAX/FIT
    RK=.0
    56J=-1.
    UY11=(.0,.0)
    DY12=(.0..)
    0 \times 21 = (-0..0)
    CY22=(.C,.C)
    DO 100 J=1.JP
    C=SG3+3.
    If(Jatúal)C=1.
    If(Jatu-JP1C=1.
    CALL KHANKIRK, KKF11
    SK=RK#SAM/SAL
    TK=RK*SAP/SAL
    C1=COS(SK)
    C2=CUS(DK1-5k)
    S1=SIN(DK2-TK)
    S2=SINITK)
    FUN=C*RKH1
    UY11=0Y11-FUN*C1*51
    UY12=0Y12-FUN*C1*52
    UY21=0Y21+FUN*C2#S1
    JY22=UY22+FJN*C2*S2
    56J=-56J
    FK=RK+UKK
100 CUNTINUE
    B=SAP *URK#D
    Y11=Y11+3*0Y11
    Y12=Y12+3*0Y12
    Y21=Y21+8*0Y21
    Y22=Y21+8*DY12
    PH=PH+UPH
    SGI = - SGI
200 CONTINUE
    B=UPH/9.
210 CONTINUE
    CNT=15./SDK1/SUK2
    S11=CNT*(CAL*S11+B*Y11)
    S12=CNT*(CAL*S12+B*Y12)
    S21=CNT * (CAL*S21+B*Y21)
    $22=CNT*(CAL*$22+B*Y22)
    RETURN
    ENU
```

Subroutine SMM3

```
SUBROUTINE SMM3(X1.Y1.X2.Y2.X3.Y3.X4.Y4.
   2DK1.DK2.S11.S12.S21.S221
    COMPLEX $11.512.521.522
    COMPLEX ET1, ET2
    COMPLEX HO.H1
    DIMENSION CC1(3C), SS1(30), CC2(30), SS2(3C)
    511=(-0--01
    512=(.0,.0)
    S21=(.0..0)
    S22=(.0..0)
    C8ET=(x2-x1)/0K1
    S8ET=(Y2-Y1)/DK1
    XA=(X3-X1)*CBET+(Y3-Y1)*SBET
    XH=(X4-X1)*CHET+(Y4-Y1)*SBET
    YA=-(X3-X1)*SBET+(Y3-Y1)*CbET
    YB=-(X4-X1)*SBET+(Y4-Y1)*CEET
    CAL = (XB + XA)/UK2
    SAL=(YB-YA)/UK2
    .00001=NIMS
    X = X \Delta
    Y = Y \Delta
    DU 40 J=1,2
    YS=Y**2
    XP = 0
    00 35 I=1.2
    0x = x - xP
    R=SWRT(DX**Z+YS)
    IF(R.LT.RMINJRMIN=R
    XP=DK1
35 CONTINUE
    A=XA+DK2 #CAL
    Y=YA+UK2*SAL
40 CUNTINUE
    ISS=4.*DK1/RMIN
    155=2*(155/2)
    1F(ISS.LT.2) ISS=2
    IF ( 155. GT. 10) 155=10
    FSS=ISS
    150 = 155 + 1
    US=UK1/FSS
    ITT=4. #DK2/RMIN
    ITT=2*(ITT/2)
    IF(ITT-LT-2)ITT=2
    IFILIT.GT.10)ITT=10
    FTT=ITT
    ITQ=ITT+1
    DT=DK2/FTT
    XP=.0
    56N=-1.
    00 50 I=1,IS4
    C = SGN + 3.
    IF(I.Ew.1)C=1.
    IF (Io Ewo ISw) C= 1.
    CC1(I)=C*CCS(DK1-XP)
    SS1(I) = C*SIN(DK1-XF)
    CC2(II=C*CUS(XP)
    SSZ(II=C*SIN(XP)
    SGH=-SGN
    xP=XP+uS
50 CONTINUE
```

```
DX = DT *CAL
    DY=UT *SAL
    X = X \Delta
    Y=YA
    TK=.0
    SGJ =- 1.
    DC 2G0 J=1.ITQ
    D=$GJ+3.
    IF(J.Ev.1)D=1.
    If (John (Tw) D=1.
    CT1=U*SIN(DK2-TK)
    CT2=D+SI4(TK)
    xP=.O
    YS= Y**2
    ET1=(.0,.0)
    ET2=(.0,.0)
    DU 100 I=1, ISQ
    DELX=X-XP
    RK=SWRT(DELX**2+YS)
    ROT= ( UELX*CAL+Y*SAL)/RK
    C1=CC1(I)
    SI=SSI(I)
    C2=CC2(I)
    S2=SS2(I)
    CALL HANK(RK, HO, H1)
    ET1=ET1-(S1*H0*CAL+C1*H1*ROT)
    ET2=ET2-(S2*H0*CAL-C2*F1*RUT)
    XP=XP+US
100 CUNTINUE
    $11=$11+CT1*ET1
    $12=$12+CT2*ET1
    S21=S21+CT1*ET2
    $22=$22+CT2*ET2
    $6J=-$6J
    TK=TK+uT
    X=X+DX
    Y=Y+DY
200 CONTINUE
    CST=(15./9.)*0S*DT/SIN(UK1)/SIN(DK2)
    $11=CST*511
    S12=CST*$12
    S21=0ST*S21
    522=CST+S22
    RETURN
    END
```

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Subroutine HANK

```
SUBPOUTINE HANK(X+H+H1)
   COMPLEX H.H1
   DATA TSP/.63662/
    X = \Delta B S (X)
    IF(X=GT=3=1G0 TC 100
    XLN=ALUG (a5*X)
    R1=x/3.
    R2=21 #R1
    R4=R2≠R2
    R6=34*R2
    IF(R1.GT..1)GC TC 50
    a=-. 16387*R6+1.26562*R4-2.25*R2+1.
    Y=. 253001*R6-. 7435C4*R4+.605594*R2+.367467
    Bl=x*(-.3954296-1*R6+.210936*R4-.5625*R2+.5)
    Y1=(-1.31648*R5+2.16827*R4+.221209*R2-.63652)/X
    Y=Y+TSP*XLN*B
    Y1=Y1+TSP*XLN*81
    H=CMPLX(B,-Y)
    H1=CMPLX(B1.-Y1)
    RETHRN
50 CUNTINUE
    28=R4*P4
    R10=R6*R4
    R12=R6*R6
    B=. 21E-3*R12-. 39444E-2*R10+.444479E-1*R8
   2-.316387*R6+1.26562*R4-2.25*R2+1.
    Y=-.24846E-3*R12+.427%15E-2*R10-.426121E-1*R8
   2+.253001*R6-.743504*R4+.605594*R2+.367467
    B1= X*(.1105E-4*E12-.31761E-3*R10+.443319E-2*R8
   2-.3954296-1*R6+.210936*R4-.5625*R2+.51
    Y1=(.276736-2*812-.4009766-1**10+.312395*88
   2-1.31648#P6+2.16E27#R4+.221269#k2-.636E21/X
    Y=Y+TSP*XL V*8
    Y1=Y1+FSP*XLN*81
    H=CMPLX(B,-Y)
    H1=CMPLX(B1-Y1)
    RETURN
100 CONTINUE
    Sw=SJRT(X)
    R1=3./X
    R2=R1*R1
    R3=R1*P2
    R4=R2*R2 .
    R5=R3*R2
    R6=R3*R3
    f = 14476E-3*R6- 72805E-3*R5+ 137237E-2*R4
   2-.95126-4*R3-.552746-2*R2-.776-0*R1+.797685
    T=.13558E-3*R6-.29333E-3*R5-.54125E-3*R4
   2+.262573E-2*R3-.3954E-4*R2-.41664E-1*R1-.785398+X
    B=F*COS(T)/SW
    Y=F*SIN(T)/SW
    F=-.20035E-3*R6+.113653E-2*R5-.249511E-2*R4
   2+.17105E-3*R3+.165967E-1*K2+.156E-5*R1+.797885
    T=-.29166E-3*R6+.79824E-3*R5+.74348E-3*R4
   2-,637879E-2*R3+,565E-4*R2+,124996*R1-2,35619+x
    B1=F*CUS(T)/SW
    Y1=F*SIN(T)/Sw
    H=CMPLX(B.-Y)
    H1=CMPLX(81,-Y1)
    RETURN
    END
```

Subroutine XHANK

```
SUBRIGHTINE XHANK (X.XH)
    COMPLEX XH
    COMPLEX G(57)
    DATA GA
               , 0.636620),( 0.004994, 0.645895).( 0.019900, 0.664765).
   . 1 0.0
   .( 0.344496, C.6879311, ( C.678411, C.712349), ( 0.121134, 0.735736),
   . ( U.172020, 0.756234), ( C.23C297, C.772274), ( 0.295073, 0.782515),
   . ( U. 365354. O. 785814; ( C. 44CUSO, C. 781213), ( O. 517992, O. 767932),
   . ( 0.397946, U.745364), ( U.678630, O.713C76), ( O.758726, O.67C806),
   • ( 0.836904, 0.618463), ( 0.911833, 0.556125), ( 0.982200, 0.484035),
   · ( 1.046729, 0.402597), ( 1.104198, 0.312372), ( 1.153449, 0.214066),
   .(1.195415, 0.108527),(1.223118,-C.003272),(1.241706,-0.120236),
.(1.248495,-C.241172),(1.242755,-0.364794),(1.224128,-0.489743),
.(1.192324,-0.614606),(1.147186,-C.737925),(1.088740,-0.858224),
   • ( 1.017176, -0.974021), ( C. 932857, -1.083849), ( 0.836299, -1.186275),
   .( J.728191,-1.279913), ( 0.609559,-1.363450), ( 0.480324,-1.43565E),
   .( 0.345678,-1.495409),( 0.199189,-1.541694),( 0.048722,-1.573635),
   .(-U.106248,-1.590497),(-0.2c4170,-1.591702),(-0.423418,-1.576836),
   .(-C.582314,-1.545654),(-C.739152,-1.498C94),(-0.8922C9,-1.434272),
   \{-1, 039768, -1, 354489\}, \{-1, 180140, -1, 259230\}, \{-1, 311676, -1, 149158\},
   • (-1.432796,-1.025115), (-1.542301,-0.888112), (-1.637894,-0.739318).
   .(-1.719192.-C.380069),(-1.784758,-C.411793),(-1.833591,-0.236107),
   .(-1.854861,-C.054691),(-1.877910, G.130665),(-1.872264, 0.318107)/
    X=ABS(X)
    IF(X.GT.5.6)G0 TO 100
    Y=13.*A
    J=Y+1.5
    IF(J.LT.2)J=2
    IF(J.GT.551J=56
    1-L = ML
    JP = J + 1
    FJ=J
    YJ=FJ-1.
    UY - Y = U
    UT=U/2.
    C = \cup T * ( \cup - 1 )
    U=JT * ( J+1. )
    E=1.-0**2
    XH = C \times G(JM) + D \times G(JP) + E \times G(J)
    RETURN
100 CONTINUE
    Sw=SwRT(X)
    R1=3./X
     R2=R1*R1
    R3=R1*A2
    R4= R2 * R2
    R5=R3*R2
    R5=R3*R3
    F=-.20033t-3#R6+.113653t-2#R5-.249511E-2#R4
   2+.171056-3*R3+.1659576-1*k2+.1566-5*R1+.797885
    T=-.29166E-3#R6+.79824E-3#R5+.74348E-3#R4
   2-a657879E-2*k3+a565E-4*R2+a124996*R1-2a35619+X
     Bl=F ≠COS(T) *Sw
     Y1=F*SIN(T)*SW
     AH≖CMPLX(B1.-Y1)
    RETURN
    END
```

Subroutine CFF

```
SUBROUTINE CFF (XA, YA, XB, YB, DK, CPH, SPH, EP1, EP2)
    COMPLEX EJA, EJB, EP1, EP2, SQJ, CST
    SQJ=30.*(1.,1.)/1.414214
    XAB=XB-XA
    YA8=YB-YA
    CA=XAB/DK
    C8=YAB/DK
    G=CA*CPH+C8*SPH
    P=C8*CPH-CA*SPH
    GK=P**2
    EP1=(.0,.0)
    EP2=(.0..0)
    IF(GK-LT-.001)GC TO 200
    A=XA*CPH+YA*SPH
    B=XB*CPH+YB*SPH
    EJA=CMPLX(COS(A),SIN(A))
    EJB=CMPLX(COS(E),SIN(B))
    SDK = SIN(DK)
    CDK = COS (DK)
    SGD=SIN(G*DK)
    CGD=CDS (G*DK)
    CST=SWJ/(P*SDK)
    EP1=CST*EJA*CMPLX(CDK-CGD,G*SDK-SGD)
    EP2=CST+EJB+CMPLX(CDK-CGD,SGD-G+SDK)
200 CONTINUE
    RETURN
    END
```

Subroutine SKETCH

```
SUBROUTINE SKETCH(XCCOR.YCGCR.NPLOT.HPAT.NSKIPS.NSYMB.NSIZES.
  1 IUNITS)
   COMMON /BLK1/ KNTPLT.ANEW.YNEW
   DIMENSION XCCOR(1). YCBOR(1)
   KNTPLT=KNTPLT+1
   WRITE(6,305) KNTPLT
   IF(KNTPLT.EG.1) 60 TO 3
   CALL CALPET (XNEW YNEW -3)
 3 CONTINUE
   XMAX=0.
   YMAX= 0.
   UD 4 I=1, NPLCT
   ATST=XCOUR(I)
   XATST=ABS(XTST)
   TELXATST-GE-XTST) XMAX=AATST
   YTST=YCOCR(I)
   YATST = ABSLYTST )
   IF(YATST.GE.YMAX) YMAX=YATST
 4 CONTINUE
   Z MA X = X MA X
   IF(YMAX.GE.XMAX) ZMAX=YMAX
   SCALE=ZMAX/(C.2*HPAT)
   HALF=0.2*HPAT
   CALL CALPLT (-HALF, -HALF, 3)
   CALL CALPLT (HALF .- HALF . 2)
   CALL CALPET(HALF.HALF.2)
   CALL CALPLY (-HALF.HALF.2)
   CALL CALPLT (-HALF.-HALF.2)
   CALL CALPLY (-HALF, C., 3)
   CALL CALPLT(HALF, 0.. 2)
   CALL CALPLT(O.,-HALF,3)
   CALL CALPLT(0. , HALF, 2)
   CALL CALPLT (0.,0.,3)
   XCOOK (NPLUT+1) =0.
   XCOOR (NPLOT+2) =SCALE
   YCGCR(NPLOT+1)=0.
   YCOUR (NPLOT+2) = SCALE
   CALL LINPLT (XCCOR, YCOCR, NPLCT, 1, NSKIPS, NSYME, NSIZES, O)
   IFINSKIPS-LT-01 GO TO 5
   STX=XCOOR(NPLOT)
   STY=YCOOR(NPLOT)
   STX1=XCOOR(1)
   STY1=YCOOR(1)
   CALL CALPLT(STX,STY,3)
   CALL CALPLT(STX1.STY1.2)
 5 CONTINUE
   HGT=U=015*HPAT
   XSC=-HALF
   YSC=-HALF+0.07 *FPAT
    IF(IUNITS.Ey.2) GD TO 6
   CALL NOTATE(XSC.YSC.HGT.15HSCALE, CM/IN = +0.+15)
    XSC=XSC+(90./7.) +HGT
   GC TO 7
 6 CALL NOTATE(XSC, YSC, HGT, 16HSCALE, WVL/IN = ,0.,16)
   XSC=XSC+(96./7.)*HGT
 7 CALL NUMBER (XSC, YSC, HGT, SCALE, 0., 4)
   CALL CALPLT(0.,0.,3)
    XNEm=0.8+HPAT
    YNEW=0.
    RETURN
305 FORMAT(25H +++++++++++ KNTPLT= 13)
    ENU
```

Subroutine DBPLOT

SUBROUTINE UBPLCT(HPAT,KI,ISK1PP,ISIZEP,
1 THTA,AEPH1,AEY,PEPH1)

```
*************
Ċ
      * PURPOSE TO USE CALCOMP EQUIPMENT TO PLOT MAGNITUDE AND PHASE
¢
                 VARIATIONS OF A SINGLE COMPONENT OF THE RADIATION
Ċ.
                 FIELL AS A FUNCTION OF ANGLE
C
Ċ
      * INPUT DATA HPAT=5.*(RADIUS OF POLAR PLOT, INCHES)
Č
                           FOR TYPE 400 PAPER, (HPAT-LE-10-)
C
                           FOR TYPE 300 PAPER. (HPAT-LE-28-)
C
                           TO OBTAIN POLAR PLOTS IDENTICAL TO PATTERN
C
Ċ
                           RECCRIBER POLAR PLUTS. USE HPAT=18.75
                    KI=NUMBER OF POINTS TO BE PLOTTED
C
                    ISKIPP=1 FUR SYMBUL EVERY DATA POINT.
C.
                           2 FOR SYMBOL EVERY OTHER DATA POINT, ETC.
Ċ
                    ISIZEP=1 FOR SMALLEST SIZE SYMBOL, 2 FOR MEDIUM
Č
                           SIZE SYMBOL. AND 3 FOR LARGEST SIZE SYMBOL
C
                    THTA=ARRAY CONTAINING THE ANGULAR VALUES (DEGREES) *
C.
                         FUR WHICH THE FIELD IS TO BE PLOTTED. FOR A
C
                         COMPLETE POLAR PLOT, THTA(1)=0., THTA(KI)=360.*
C
                    AEPH1=ARRAY OF MAGNITUDE VALUES (DB) CORRESPONDING *
Ċ
                          TO ANGULAR VALUES IN THTA ARRAY. VALUES *
MUST LID BETWEEN C DB (MAX) AND -40 DB (MIN)*
С
C
                    PEPHI=ARRAY OF PHASE VALUES (DEGREES) CORRESPONDING*
C
                          TO ANGULAR . VALUES IN THTA ARRAY. VALUES
C
                          MUST LIE BETWEEN 18C. (MAX) AND -180. (MIN)
٤.
      * AEY IS AN ARRAY USED FOR INTERMEDIATE CALCULATIONS
C
C
                      THE ARRAYS THTA, AEPHI, PEPHI, AND AEY MUST
Ĉ
      * RESTRICTIONS
                      BE DIMENSIONED AT LEAST AS LARGE AS THE VALUE
C
                      (KI+2) IN THE CALLING PROGRAM. CCMMCN /BLK1/
C
                      KNTPLT . XNEW . YNEW MUST ALSO APPEAR IN CALLING
C
۲.
                      PROGRAM
С.
      *********
C
      COMMON /BLK1/ KNTPLT, XNEM, YNEM
      DIMENSION THTA(1), AEPHI(1), AEY(1), PEPHI(1)
      PI=3.141592653589793
      RDN=PI/180.
      KNTPLT=KNTPLT+1
      WRITE(6,305) KNTPLT
      IF(KNTPLT.Eq.1) GO TU 191
      CALL CALPLTIANEN, YNEM, -31
  191 CONTINUE
      STX=0.2075*HPAT
      STY=0.00375*HPAT
      STH=5.015* HPAT
      CALL NUMBER (STX, STY, STF, C., O., -1)
      STX1=STX+J.01286#HPAT
      STY1=STY+G.010*HPAT
      STH1=J.006*HPAT
      CALL NUMBER (STX1, STY1, STF1, C., O., -1)
      STY =- 0.01875#HPAT
      CALL NUMBER (STX, STY, STH, 360.,00, -1)
      STX1=STX+0.03657*HPAT
      STY1=STY+0.010#HPAT
```

```
CALL NUMBER (STX1.STY1.STF1.0..0...1)
    5TX=-0.01071*HPAT
    STY=0.2075*HPAT
    CALL NUMBER (STX.STY.STF.9C..C..-1)
    STA1=STX+G.02571*HPAT
    STY1=STY+Da010*HPAT
    CALL NUMBER (STX1.STY1.STH1.Co.O.,-1)
    STA=-0.2495*HPAT
    STY=-0.0075*HPAT
    CALL NUMBER (STX.STY.STF.180..0.,-1)
    STX1=STX+0.03857*HPAT
    STY1=STY+0.010*EPAT
    CALL NUMBER(STX1.STY1.STF1.0.,U.,-1)
    STX =- 0. 01714*HPAT
    STY =- C. 2225*HPAT
    CALL NUMBER (STX, STY, STF, 270, 00, -1)
    STX1=STX+0.03857*HPAT
    STY1=STY+O-010*HPAT
    CALL NUMBER (ST X1, STY1, STF1, 0., u., -1)
    SRHUU=25.
    SRHO= SRHOU+RUN
    SRAD=O. 2025*BPAT
    STX=-SRAD*SIN(SRFC)
    STY=SRAU*C US (SPHO)
    CALL NUMBER (STX, STY, STH, O., SRHOD, -1)
    CALL NOTATE(STX, STY, STH, 4H CB, SRHOD, 4)
    SRAU=0.1525*HPAT
    STX=-SRAD*SIN(SREQ)
    STY=SRAD*COS(SRED)
    CALL NUMBER (STX, STY, STF, -10., SRHOU, -1)
    SRAU=C. 1025*HPAT
    STX = - Sk AU + SIN( SKHO)
    STY=SRAD*CUS(SRHO)
    CALL NUMBERISTX, STY, STF, -20., SRHOD, -1)
    SRAD=C.0525*HPAT
    STX=-SRAJ*SIN(SRHG)
    STY=SRAU*CCS(SRHC)
    CALL NUMBER (STX, STY, STH, -30, , SRHOD, -1)
    HRAD=C. 2*HPAT
    CALL CALPLT (HRAD. O. . . )
    LENU=6
    DSRHO=PI/FLOAT(LEND)
    00 195 L=1, LEND
    LM1=L-1
    DSRHU1=OSRHC*FLCAT(LM1)
    SUX1=HRAD*COS(USRHOI)
    SDY1=HRAD#SIN(OSRHOL)
    CALL CALPLT (-SCX1,-SDY1,2)
    DSRH02=DSRHO*FLCAT(L)
    SUX2=HRAD*COS(USRHO2)
    SDY 2=HRAD*SIN( DSRHC2)
    CALL CALPLT(SOX2,SEY2,3)
193 CUNTINUE
    CALL CALPLT(C. . C. . 3)
    KEND=4
    SURU= HR AD/FLUAT (KEND)
    DU 195 K=1, KEND
    XXO=SURU*FLUAT(K)
    CALL CIRCLE(XXU,0.,0.,36C., xxC, xxO,3)
195 CONTINUE
```

```
TARHY COS=1 I TABG
    00 221 KX≃1•KI
    THTAL =THTALKX 1 *RON
    AEPH2 = AEPH1 (KX)+40.
    AEPHI(KX)=(AEPH2/UEPIN)*COS(THTA1)
    AEY(KX)=(AEPH2/LBPIN)*SIN(THTA1)
221 CONTINUE
    AEPHI(KI+1)=0.
    AEPH1(KI+2)=1.
    AEY(KI+1)=0.
    AEY(K[+2)=1.
    CALL LINPLT (AEPHI, AEY, KI, 1, ISKIPP, 1, ISIZEP, C)
    XNEW=-U. 2*HPAT
    YNE = -0.52 + HPAT
    KNTPLT=KNTPLT+1
    WRITE(6.305) KNTFLT
    CALL CALPLT(XNEW, YNEW, -31
    DSTCE=Q.4*HPAT
    UUVV1=360./DSTCE
    XTMAJ=J<sub>■</sub>25*BSTCE
    . PLANTA = NIMTX
    TARH*SC.C=HTZ
    CALL AXES(0.,0.,0.,USTCE,0.,UDVV1,XTMAJ,XTMIN,14HTHETA, DEGREES,
   1 STH - 141
    DSTCE=0.3*HPAT
    DCVV2=36C./OSTCE
    ATMAJ=0.25#DSTCE
    *PYLAMTX=MIMTX
    CALL AKES(0.,0.,90.,USTCE,-180.,DUVV2.XTMAJ.XTMIN.
   1 14HPHASE, DEGREES, STH, 141
    XSGRU= (0.4*HPAT)/12.
    YSURU= (0.3+HPAT1/4.
    CALL GRID(O., C., xSGRU. YSGRD. 12.4)
    THTA(KI+1)=0.
    THTA(KI+2)=DDVV1
    PEPH1 (KI+1) =-180.
    PEPHI(KI+21=DUVV2
    CALL LINPLTITHTA, PEPHI, KI, 1, ISKIPP, 1, ISIZEP, 0;
    ANEW= O. 8*HPAT
    YNEW= 0.62*HPAT
    RETURN
305 FURMAT(25H +++++++++++ KNTPLT= 13)
    ENÓ
```

APPENDIX B

NUMERICAL EXAMPLE

The purpose of this appendix is to present a numerical example which illustrates the application of the digital computer programs presented and described in appendix A. The example is the computation of the roll-plane radiation patterns for case 3. In the example, the frequency range is the interval of 5.250 GHz to 14.000 GHz, with an increment of 1.750 GHz. Consequently, FMCO = 5250., FMCD = 1750., and FMCF = 14000. The two annular slots for case 3 require four equivalent narrow axial slots; consequently, NPORT = 4.

The cross-section profile, shown in figure 8, is initially described, as shown in figure 13, by 55 coarse points; thus, NPTIN = 55. In figure 13, the numbers between any two consecutive points specify the number of segments to occur between the two points by subroutines SPLFIT and SPFIT2. As noted in appendix A, the number of points generated between a pair of consecutive points is one less than the number of segments that occur. In figure 13, coarse point 27 appears to lie in an approximately linear region of the profile; consequently, point 27 is chosen as the point for initiation of the spline fit procedure; that is, ISTART = 27. Other input data for the example are that MADM = 0, KI = 361, KWRT1 = 1, KWRT2 = 1, KWRT3 = 1, ISKIPP = 15, ISIZEP = 1, and HPAT = 18.75.

Input Data Cards

A listing of the input data cards for the numerical example is on the following page. The first card contains, in sequence, the values NPTIN, ISTART, NPORT, MADM, KI, KWRT1, KWRT2, KWRT3, ISKIPP, ISIZEP, HPAT, FMCO, FMCD, and FMCF.

The next four cards are cards which specify the port index I, the coarse point index JVGS(I), and the complex voltage strength VGS(I) for each of the four equivalent narrow axial slots which excite the configuration. The first two cards apply for the annular slot located at B_1 (see figs. 1 and 8) and the second two cards apply for the annular slot located at B_2 .

The next 55 cards specify identifying integers and the X- and Y-coordinates, in cm, of the 55 coarse points. The first of these cards is for coarse point 1, the second for coarse point 2, and so forth. In particular, on the Ith card are the identifying integer IGNORE, the X-coordinate PNTIN(I,1) in cm, and the Y-coordinate PNTIN(I,2) in cm of the Ith coarse point.

APPENDIX B - Continued

| | | 1.000 | 0. | | | | |
|------------------------------|--------------------|----------------|----|---|-------|-------|------------------|
| 5 I | 3 9 - | 1.000 | 0. | | | | |
| | 29 - | 1.000 | 0. | | | | |
| a | 35 | 14000 | 0. | | | | |
| | | | | | | | |
| | | - - | | | | | |
| | | | | | | | |
| 1 | 9.610 | 0.000 | | | | | - |
| 3001 | 9,388 | 0.797 | _ | | | | |
| 2002 | 9.369 | | | | | | |
| 3003 3004 | 9.350 9.257 | 0.934 1.25B | | | | | |
| 700 | 9.165 | 1.600 | | | | | |
| 1006 | 9.073 | : •932 | | | | | |
| 3007 | 8.960 | 2.266 | | | | | |
| 2008 | 8.761 | | | | _ | | |
| ימטר. a | 9.942 5.720 | 3.200 | | | | - | |
| | 7.680 | 6.210 | | | | | |
| 18 | 7.190 | 8 - 1 0 2 | | | | | |
| 25 | 6.310 | 9.936 | | | | | |
| . 76 | 5.200 | 11.500 | | | | | |
| 36 | 2.000 2.580 | 12.810 | | | | | |
| 34 | 1.300 | 14.650 | | | | | |
| 41 | 0.000 | | | | | | |
| 45 | -1.300 | 14-635 | | | | | |
| 48 | -24560 | | | | | | |
| <u>52</u> | -4.000 | 12 is 10 | | - | | - | |
| <u>56</u> | -5.200 -6.310 | 11.500 | | | | | |
| 64 | -7-190 | | | | | | |
| 68 | -7.880 | 6.210 | | | | | |
| 74 | -8.720 | 3.200 | | | | | |
| 7011 | -B.942 -B.961 | | | | | | |
| 3015 | -6+761 | 2.334 | | | | | · - - |
| | | | | | | | |
| | | | | | | | |
| 3013 | -8.980 | 2.266 | | | | | |
| 3014 | -9.073 | 1.932 | | | | | |
| 3015 | -9-165 | 1.500 | | | | | |
| 701 <u>6</u> 701 <u>7</u> | -9.257 -9.350 | | | | | | |
| 3018 | -9.350 | | | | | | |
| 3019 | -9.388 | | | | | | |
| 81 | -9.610 | 0.000 | | | | | |
| 86 | -10,500 | -3.200 | | | | | |
| 91 94 | -10.800 -10.600 | | | | | | |
| 97 | -9.500 | | | | | | |
| Ion | -8-000 | | _ | | | | |
| 104 | -6,000 | -7:050 | - | | | | |
| 168 | -4.000 | -7.265 | | | | | |
| 112 | -2.000 0.000 | | | | | | |
| 116 | 8.000 | | | | | | |
| 124 | 4.000 | | | | | | |
| 128 | 6.000 | -7.050 | | | | | |
| 132 | 8.000 | -6:540 | | | | | |
| 135 | 9,500 | | | | - | | , |
| 138 | 10.600 | | | | | | |
| 141 | 10.800 | | | | | | |
| 151 | 9.610 | | | | | | |
| | _ : | | | | | | |

The last two cards contain the IDIVD array data. There are 54 integers (NPTIN-1) which constitute the IDIVD array data. The first value, IDIVD(1) = 3, indicates that three tegments occur between coarse points 1 and 2. Thus, subroutines SPLFIT and SPFIT2 generate two points between coarse points 1 and 2. The second value IDIVD(2) = 1 indicates that only one segment occurs between coarse points 2 and 3. Consequently, suboutines SPLFIT and SPFIT2 generate no additional points between coarse points 2 and 3. I similar interpretation is given for the remaining IDIVD array values. The sum of the DIVD array values is 166; thus, the IDIVD array data cause 166 actual points to be produced for use in subroutine TESLOT. A plot of the 166 points is given in figure 8. The lew indices for the four equivalent narrow axial slots are also given in figure 8 and in able I.

Execution

After the input data cards are read by the main program, execution occurs. The printer output data follow:

```
ISTART=27
NPTIN=55
IDIVL ARRAY VALUES
3 1 1 1 1 1 1 1 1 3 6 4 4 4 4 4 4 4 4 4 4 4 4 4 6 3 1 1 1 1 1 1 1 3 7 3 3 3 3 4 4 4 4 4 4 4 3 5 3 3 7
                                    KI KWRTI KWRTZ KWRT3 ISKIPP ISIZEP HPAT FMCD FMCD FMCF
                            мавм
NPOINT MANPT NPORT MAPORT
                                                                      1 18.75 5250.0 1750.C140C0.0
                                                               15
         170
++++++++++++ KNTPLT=
           LOCATION OF POINTS
                               Y, CM.
    POINT
                X. CM.
                               0.0000
                 9.6103
                 9.5365
                               . 2658
                                .5313
                 9.4620
                                .7970
                 9.3880
                 9.3690
                                . 3660
                                40 دو.
                 7.3500
                               1.2640
                 9. 2570
                9.1650
                               1.6000
       q
                 9.0730
                               1.9323
       10
                 9.9800
                               2.2660
                8.9610
                               2.3340
                 8.9420
                               2-4030
                 8.6583
                               2.6627
                 8.7943
                               2.9343
                               3.2000
                 8.7203
                               3.7027
                a.5832
       17
                8.4487
                               4.2060
       18
                8.3137
                               4.7092
                B. 1757
                               5.2114
                               5.7120
                 B. 032J
       21
                 7.8803
                               6-2100
       22
                 7.7232
                               6.6879
                               7.1625
                 7.5562
                               7-6333
       24
                 7.3785
                               8.1000
       25
                 7.19CJ
                 6.9911
                               8.5002
       26
                 6.7794
                                9.0146
       27
                               9.4617
                 6.5530
                               9.9000
                 6.3100
                              10.3104
                 6.0566
                 5.7866
                              10.7223
                 5,5008
                              11.1171
                 5.2000
                              11.5000
                 4a 9135
                              11.63E7
       4د
                 4.6163
                               12.1687
                 4.3118
                              12.4518
       35
                 4.0000
                               12.0100
       3.8
                 3,6560
                               13.1390
       39
                 3.3213
                               13.4561
       40
                 2.9510
                               13.7526
                 2.5800
                               14.0200
                 2,2770
                               14.197c
       42
                 1.9014
                               14.35:7
                 1.6351
                               14.4679
                 1.3000
                               14.6000
                  . 9802
                               14.0844
       46
                  .6559
                               14.7473
                  -3287
                               14.7665
                 0,0000
                               14.8000
       F0
                 287د --
                               14.7865
                 -a of 59
                               14.7473
                 -.9804
                               14.6844
```

```
E 3
           -1-3003
                            14.6006
-4
           -1.6351
                            14.4879
 5.5
           -1.9614
                            14. 3537
80
           -2.2770
                            14-1976
                           14.0200
           -2.5800
57
58
           -2.961C
                           13.7526
 59
           -3.3213
                            13.4561
60
           -3.6660
                            15,1390
           -4.0300
                          12.8100
61
           -4.3118
                            12.4918
62
           -4.6169
                            12.1687
64
           -4.9135
                            11.8387
65
           -5.2000
                           11.5000
66
           -5.5007
                            11.1171
67
           -5. 7865
                            10,7223
           -6.0565
                            10.3163
68
69
           -6.3100
                             9,9000
 70
           -6.5531
                             9.4617
 71
           -6.7797
                             9.0147
           -6.9914
                             8.5603
 72
 73
           -7.1900
                             8.1000
           -7.3779
 74
                             7.0331
 75
           -7.5551
                             7.1622
 76
           -7.7223
                             6.6876
 77
           -7.8800
                             6.2100
           -8.0343
                             5.7127
 79
           -8.1808
                             5-2129
 80
           -8.3210
                             4.7114
 81
           -8.4566
                             4.2084
 82
           -8.5891
                             3.7644
 83
           -8.7200
                             3.2000
 84
           -b. 7944
                             2.9345
 85
           -8.8685
                             2.6688
 56
           -8.9420
                             2.4030
 87
           -8.9610
                             2.3340
 88
           -8.9800
                             2.2660
 89
           -9.0730
                             1.9320
 90
           -9.1650
                             1.6000
 91
           -9.2570
                             1.2680
 92
           -9.3500
                             9340
           -9.3690
                              . Sc & C
 94
           -9.3380
                              .7970
 95
           -7,4620
                              5313
           -9.5366
 96
                              .2658
                             0.0600
 97
           -5.6100
 98
           -9. 7303
                             -.4506
                            -.5183
 99
           -9.8457
100
           -5.9607
                            -1.37.61
101
          +10.0/93
                            -1.8371
102
          -1ŭ. 2059
                            -2.2944
103
          -10.3447
                            -2.7490
104
          -10.5000
                            -3.20CO
105
          -10.6277
                            -3.5441
106
          -1u.7377
                            -3.8920
107
          -10.8000
                            -4.25CU
          -10.7947
                            -4.5782
108
          -10.7281
                            -4.9010
109
          -10. 2000
                            -5.20L0
110
          -10.3086
                            -5.5685
111
           -9.9278
                            -5.8629
112
           -9.5000
                            -6.1000
113
           -9.0066
                            -6.3163
114
115
           -8,5063
                            -6.4926
            -8.0000
                            -6.6400
116
           -7.5065
                            -6.76:3
117
            -7-0077
                            -6.3718
118
            -6.5050
                            -6.9669
119
            -6.0000
                            -7,05CG
120
121
            -5.5011
                            -7.121c
            -5.0014
                            -7.1635
122
123
            -4.5011
                            -7.2561
                            -7.2860
            -4.0000
124
145
            -3.5006
                            -7.31:7
125
            -3,0007
                            -7.3459
            -2.5005
                            -7.3731
127
128
            -2.0000
                            -7.4000
129
            -1.5000
                            -7.4260
130
            -1.0000
                            -7.4538
131
             -.5000
                            -7.4727
             3.0000
                            -7.48CO
152
```

| 133 134 135 136 137 133 139 141 142 143 164 145 | .5000 1.0003 1.5000 2.000 2.5005 3.0007 3.5006 4.0000 4.5011 5.0014 5.5011 5.0000 | -7.4727 -7.4538 -7.4260 -7.4766 -7.3731 -7.3459 -7.3157 -7.2860 -7.2381 -7.1216 -7.0560 -6.9669 |
|--|--|--|
| | | |
| 142 | | |
| 143 | | |
| 164 | 5 . 0J00 | |
| 145 | 6,5050 | |
| 146 | 7 . თა77 | -6.8718 |
| 147 | 7. 5065 | -6.7623 |
| 148 | 8,0000 | -6.64CŬ |
| 149 | 8,5068 | -c.4926 |
| 150 | 9.0066 | -6.216= |
| 151 | 9,5000 | -5. 1CGO |
| 152 | 9.9278 | -5.9629 |
| 153 | 10.3086 | -5.5685 |
| 154 | 10.6303 | -5.20uJ |
| 155 | 10.7281 | -4.9010 |
| 156 | 10.7947 | -4.5782 |
| 157 | 10.8000 | -4.2500 |
| 158 | 40.7377 | -5.8920 |
| 159 | 10.6277 | -3.5441 |
| 160 | 10.5000 | -3.2000 |
| ltl | 1 C. 3447 | -2.7450 |
| 162 | 10.2059 | -2.2944 |
| 163 | 10.0793 | -1.8371 |
| 164 | 9.9607 | -1.3781 |
| 165 | 9. 8457 | 9183 |
| 166 | 9.7303 | 4526 |

FREWUENCY= 5250.0 MHZ WAVELENGTH# 5.71 CM.

++++++++++++ KNTPLT = 2

NUMBER OF FOINTS DESCRIBING THE CYLINDER=106

| GEOMETR | Y OF CYLIN | DRICAL CRESS | SECTION | DRIVING POINT | |
|---------|------------|------------------------|------------------------|---------------|--------------|
| TAIGS | A, WVL. | Y, WVL. | D, AVL. | RE(V), VCLTS | IM(V), VOLTS |
| | | | | | |
| _ | | 4 4004 | 24.02 | | |
| l | 1.6818 | 0.0000 | .2483 | | |
| 2 | 1.6589 | .0465 | .0483 | | |
| 3 | 1.6559 | .093 C | .0483 | | |
| 4 | 1.6429 | 1395 | •C125 | 1 2020 | 0.0000 |
| 5 | 1.6396 | .1516 | .0124 | 1.0000 | 0.0000 |
| 6 | 1.6363 | .1635 | .0607 | | |
| 7 | 1.6200 | .2215 | .0603 | | |
| θ | 1.6039 | .2800 | .0603 | | |
| Ą | 1.5878 | .3381 | -L6C7 | | |
| 10 | 1.5715 | . 3966 | .0124 | | 0.0000 |
| 11 | 1.5602 | 4985 | .0125 | -1.0000 | 0.0000 |
| 12 | 1.5648 | · 4205 | •C483 | | |
| 13 | 1.5520 | .4670 | .0483 | | |
| 14 | 1.5349 | •5135 | C483 | | |
| 15 | 1.5260 | . 5€30 | •C912 | | |
| 16 | 1.5021 | 6480 | .0912 | | |
| 17 | 1.4785 | .7 360 | .0912 | | |
| 18 | 1.4549 | .8241 | .0912 | | |
| 19 | 1-4307 | •912C | .0911 | | |
| 20 | 1.4056 | .9996 | -C911 | | |
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| 22 | 1.3516 | 1.1704 | .0880 | | |
| 23 | 1.3223 | 1.2534 | .0881 | | |
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| 28 | 1.1468 | 1.0558 | • Ca77 | | |
| 29 | 1.1043 | 1.7525 | • Oa53 | | |
| 30 | 1.0599 | 1.8054 | • Ca5a | | |
| 30 | 2.0000 | 103027 | | | |

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| 113 | -1.6625 | | .0927 |
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| 115 | -1.4887 | -1.1362 | -0924 |
| 116 | -1.4000 | -1.1620 | • CB90 |
| 117 | -1.3136 | -1.1836 | .0093 |
| 118 | -1.2263 | -1.2026 | •G895 |
| 119 | -1.1384 | -1.2192 | .ú896 |
| 120 | -1.0500 | -1.2338 | .0862 |
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| 122 | 0753 | -1.2571 | • C8aG |
| 123 | 7877 | -1.2663 | .C880 |
| 124 | 7300 | -1.2740 | .CB76 |
| 125 | 6126 | -1.2803 | . Cd76 |
| 126 | 5251 | -1.2855 | .C877 |
| | 4376 | -1.2903 | • C877 |
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| 129 | 2625 | -1.2999 | .0876 |
| 130 | 1750 | -1.3044 | |
| 131 | 1875 | -1.3077 | • CB75 |
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| 133 | .C875 | -1.3077 | -0876 |
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| 137 | .4576 | -1.2903 | .6877 |
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| 143 | .9é27 | -1.2463 | -C582 |
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| 146 | 1.22£3 | -1.2026 | • C893 |
| 147 | 1.3136 | -1.1B36 | . 0890 |
| 148 | 1.4000 | -1.1.20 | . ú924 |
| 149 | 1.4887 | -1.1362 | •¢927 |
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| 151 | 1.6625 | -1.0675 | .0356 |
| 152 | 1.7374 | -1.0250 | .0842 |
| 153 | 1.8040 | 9745 | .0822 |
| 154 | 1.8550 | 9130 | € C5 59 |
| 155 | 1.8774 | 8577 | • C577 |
| 156 | 1.8891 | 8012 | .C574 |
| 157 | 1.8900 | 7438 | • C63£ |
| 158 | 1.8791 | 6811 | .0638 |
| 159 | 1.8598 | - 6232 | .C642 |
| 160 | 1.8375 | - 500C | .C835 |
| | 1.8103 | 4811 | .0532 |
| 161 | 1.7860 | 4015 | .0830 |
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| 163 | 1.7431 | 2412 | • C829 |
| 164 | | | .6829 |
| 165 | 1.7239 | 1607 | |
| 166 | 1.7028 | €803 | .0850 |

RADIATION PATTERN (RELATIVE)

| PHI, | ΰĒĠ. | RE(EPH) | IM(EPH) | MAG (EP+) | PHASE. DEG. |
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| | 2.0 | -3.5469106-01 | 1.6578536-01 | 3.915250E-01 | 154.95 |
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| | 4.0 | -3.641023E-01 | 1.349373E-C1 | 3.883021E-C1 | 159.67 |
| | 5.0 | -3.511091t-01 | 1.014642E-C1 | 3.654759E-01 | 163.88 |
| | 6.0 | -3.328930E-01 | 5.721147E-C2 | 3.377734E-01 | 170.25 |
| | 7.0 | -3.127275E-01 | 8.974877E-C3 | 3.128563E-01 | 178.36 |
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| | 9.0 | -2.353321E-01 | -8.286414E-02 | 2.494948E-01 | -160.60 |
| | 10.0 | -1.813504E-01 | -1.209255E-01 | 2.179722E-01 | -146.30 |
| | 11.0 | -1.251474E-01 | -1.573795E-C1 | 2.010725E-01 | -128-49 |
| | 12.0 | -5.942743E-02 | -1.868344t-Cl | 1.960579E-C1 | -107.64 |
| | 13.0 | 3.245147E-03 | -1.874396E-C1 | 1.874677E-01 | -89.01 |
| | 14.0 | 6.8806998-02 | -1.916619E-01 | ∠.036386E-01 | -70.25 |
| | 15.0 | 1. 2356246-01 | -1.80C127E-C1 | 2.183397E-01 | -55.53 |

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| 104.0 | 1.330961E+30 | 1.348571E+CO | 1.854757E+CC | 45.38 |
| 105.0 | 1.383202E+36 | 1.533482E+CO | 2.0651436+00 | 47.95 |
| 106.0 | 1.39550CE+C3 | 1.686539E+UJ | 2.189025E+00 | 50.39 |
| 107.0 | 1.3773228+00 | 1.7998c6E+00 | 2.2663928+00 | 52,58 |
| 108.0 | 1.399204E+00 | 1.751724E+00 | 2.341943E+00 | 51.38 |
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| 119.0 | -9.521476E-01 | 1.0544CzE-C1 | 9.57968CE-Cl | 173.68 |
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| 135.0 | | | | -71.1C |
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| 138.0 | 3.825383E-01 | -5.795820E-G1 | 6.944429E-01 | ÷56•57 |
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| 144.0 | 3.775354E-01 | -6.791975E-C1 | 7.7707298-01 | -60.93 |
| 145.0 | 3.062417t-01 | -6.498724E-01 | 7.184136E-C1 | -64.77 |
| 146.0 | 2. 347245E-01 | -5.972785E-C1 | b. 417455E-01 | -68.55 |
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| 158.0 | 1.6839686-01 | 2.2187556-02 | 1.698522E-C1 | 7.51 |
| 159.0 | 2.112436E-01 | -6.9362136-03 | 2.113574E-C1 | |
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| 135.0 | | | | -71.1C |
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| 146.0 | 2. 347245E-01 | -5.972785E-C1 | b. 417455E-01 | -68.55 |
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| 149.0 | 5.816610E-02 | -3.861363E-01 | 3.90486EE+01 | -81.43 |
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                         1.093960E+CO
                                        1.0971956+00
                                                                94.40
200.0
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                         1.179839E+23
                                        1-17985CE+00
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        6. Ca 8804c-01
                         1-31466CE+CO
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                                                                61.57
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                                        1.363580E+00
                         L. 1040716+00
                                                                54.07
211.0
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                         5-920473E-01
                                        1.313674F+CC
                                                                49-04
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212.0
                         F-644224F-C1
                                        1-25893PE+OC
                                                                43.3E
213.0
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        1.110980F+00
216.0
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                                        1.128482E+0C
                                                                10.10
217.0
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                         1-7029311-03
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                                                                  .00
216.0
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                                        1-186653E+00
                                                                -9-86
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                                        1.244185E+CC
                       -3-9804556-C1
                                                               -18-66
220.0
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                       -5.802468E-01
                                        i.31944CE+CC
                                                               -26-09
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                       -9.3108956-C1
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                                        1.4746118+00
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223.0
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                       -1.077213E+CG
                                        1.537273E+C0
                                                               -44.49
224. C
        1.0291212+00
                       -1.191562E+CO
                                        1-5744566+00
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                                        1.5646806+00
                                                               -58-56
227.0
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                                        1.52031 SE+0E
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228 . 0
        5.1002426-01
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                                        1.448471E+CO
                                                               -69.38
229-0
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                                        1.244665E+0C
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235.D
                                                              -150.80
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                                        1.072093E+C0
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                                                               169.86
       -1 + 3508666+00
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                                        1.32579CE+CO
                                                               149.37
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                                                               143.84
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251.0
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                                        1.158495E+0C
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255.0
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                                                                 •02
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|-----------------|------------------------|---------------------------------|--------------|-----------------|
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| 257.0 | 1.198228t+OC | -1.327117e-Ci | 1.205555E+00 | |
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| Z60.0 | 7.376110E-31 | -2.200315E-C1 | 7.697297E-01 | |
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| 263.0 | 2.135070E-04 | -1.75 to 2 (4 ti- C) | 1.756305E-01 | |
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| 267.0 | -9.061212E-01 | -6.135571E-C2 | 9.081561E-01 | -176.13 |
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| 269.0 | -1.130263t+30 | -2.87i276E-02 | 1.1306336+00 | -178.54 |
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| | | -2.996271E-C2 | 1.133319E+CC | -178.49 |
| 271.0 | -1.132923E+60 | | | |
| 272.0 | -1.C50301E+00 | -4.441637c-02 | 1.05124CE+CC | -177.58 |
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| | | | | -172.83 |
| 274.0 | -7.326467E-01 | -9.213801t-C. | 7.384176E-C1 | |
| 275.0 | -5.149802E-01 | -1.22754GE-Gl | 5.294C37E+C1 | -166.59 |
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| 277.0 | -1.32084JE-02 | -1.831267E-C1 | 1.8360246-01 | |
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| 285.0 | 1.211501E+0C | | | |
| 286.0 | 1.161828E+90 | 3.C97166E-02 | 1.162241E+00 | 1.53 |
| 287.C | 1.080298E+00 | 7.537BC5t-C2 | 1.0829256+00 | 3.99 |
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| 290.0 | 4.6224756-01 | 4.434973E-C1 | 6.3852226-01 | 43•62 |
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| 298.0 | -1.238566E+00 | 6.1622C6E-01 | 1.383392E+00 | 153.55 |
| 299.0 | -1.314922E+JO | 5.206936E-01 | 1.414264E+OC | 158.40 |
| | | | | |
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| | | | | -59.45 |
| 314.0 | 7.853640E-01 | -1.332477E+CC | 1.547212E+CC | |
| 315.0 | 9.0262C1E-01 | -1.285140£+CO | 1.568812E+00 | -54-88 |
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| 317.0 | 1.069321E+00 | -1.080866E+00 | 1.520434E+00 | |
| 318.0 | 1.1173626+00 | -9.372335E-01 | 1.4587748+00 | -39.98 |
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| 320.0 | 1.154157E+00 | -5.919422E-01 | | |
| 21.0 | 1.160858E+00 | -4.127517t-C1 | 1.232054E+00 | -19.57 |
| 322.0 | 1.1542171+00 | -2.210154E-C1 | 1.175187E+0C | -10.84 |
| | | | | 98 |
| 323.0 | 1.131211E+00 | -1.938910E-02 | 1.131377E+CC | |
| 32 4. € | 1.102414E+06 | 1.737374E-C1 | 1.116021E+00 | 8.96 |
| 325.0 | 1.060142=+30 | 3.529418E-C1 | 1.1173498+00 | 18-41 |
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| 327.J | 9.716211E-01 | 6.84377.E-Cl | 1.188453E+CO | 25.16 |
| 328.0 | 9.200\$83c+01 | E.288256E-C1 | 1.238329E+00 | 42.01 |
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| 329.0 | 9.691685E-01 | 9.54.7006-01 | 1.29C843E+00 | 47.68 |
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| 3:1.3 | 7.3811766-01 | 1.152878E+CJ | 1.379808E+CG | 56.67 |
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| 333.0 | 5.266124E-01 | 1.274368E+CO | 1.420J9CE+00 | ·63 . 82 |
| 334.0 | 5.350364E-01 | 1.307783E+CO | 1.426699E+00 | 67.00 |
| | | | | 70.37 |
| 3.5.0 | 4.7376976-DI | 1.3282C2E+CO | 1.4101708+00 | 10.51 |
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| 336.0 | 3.9737248-01 | 1.323130E+CC | 1.381F12F+00 | 73.28 |
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| 337.0 | j.125672£−01 | 1.298820E+CG | 1.335902E+CC | 76-47 |
| 338.0 | 2.2300C6c-01 | 1.25787CE+CU | 1.277485E+CC | 79.95 |
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| 340.0 | 3.6713886-02 | 1.161322E+CC | 1.161903E+CO | 88.19 |
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| 342.0 | -1.432648E-01 | 9-952867E-C1 | 1.0055488+00 | 93.19 |
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| 355.0 | -4.0691356-01 | 1.413233E-C1 | 4.307562E-C1 | 160.85 |
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| 360.0 | -3.54566dE-01 | 1.796989E-C1 | 3.975035E-01 | 153.12 |

RADIATION PATTERN IDB AND EZEMAXI

| PHI, DEG. | EPH+ ∪B+ | é/EMAX | PHASE, DEG. |
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| 84.0 | -6.345976 | . 481616 | |
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| 115.0 | -9.622184 | • 3 3 0 2 E 6 | 83.87 |
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| 245.0 | | | | |
| 246.0 -8.844767 361212 128.32 247.0 -10.565238 296264 115.07 248.0 -12.140207 247167 95.88 249.0 -12.880667 226959 69.93 250.0 -10.488038 298950 26.45 252.0 -8.779878 36.920 14.76 253.0 -7.308887 431078 4.78 254.0 -6.730121 460780 1.80 255.0 -6.356771 481018 0.2 255.0 -6.356771 481018 0.2 255.0 -6.356771 481018 0.2 255.0 -6.356771 481018 0.2 255.0 -6.356771 481018 0.2 255.0 -6.356771 481018 0.2 255.0 -6.356462 479458 -6.32 258.0 -7.219650 435529 -9.72 259.0 -8.433596 .378722 -12.75 260.0 -10.281236 306153 | | | | |
| 247.0 -10.565238 2963C4 115.07 248.0 -12.140207 247167 55.88 249.0 -12.880667 226969 69.93 250.0 -12.050987 249718 44.39 251.0 -10.488038 298950 26.45 252.0 -8.779878 36.5520 14.76 253.0 -7.308887 -451078 4.77 254.0 -6.730121 -460780 1.80 255.0 -6.356771 -481018 -02 255.0 -6.356771 -489035 81 257.0 -6.364262 -479458 -6.32 258.0 -7.219650 -435529 -9.72 259.0 -8.433596 -578722 -12.75 260.0 -10.281236 -306153 +16.61 261.0 -13.515141 -220005 -22.72 262.0 -17.701294 -130257 -37.34 263.0 -23.116000 -669655 -89.93 264.0 -16.53 | | | | |
| 248.0 -12.140207 247167 95.88 249.0 -12.880667 226959 69.93 250.0 -12.050987 244718 44.39 251.0 -10.488038 298950 26.45 252.0 -8.779878 36.920 14.76 253.0 -7.308887 431078 4.78 255.0 -6.730121 466780 1.80 255.0 -6.355771 481018 02 255.0 -6.355771 481018 02 257.0 -6.384262 479458 -6.32 258.0 -7.219650 435554 -9.72 259.0 -8.433596 376722 -12.75 260.0 -10.281236 306153 +16.61 261.0 +13.151341 220005 -22.72 262.0 -17.701294 130297 -37.34 263.0 -23.116000 6069855 +89.93 264.0 -16.533966 118386 -150.33 265.0 -13.73776G | | | | |
| 249.0 -12.880667 226969 69.93 250.0 -12.050987 249718 44.39 251.0 -10.488038 298950 26.45 252.0 -8.779878 36.5920 14.76 253.0 -7.308887 431078 4.78 254.0 -6.730121 460780 1.80 255.0 -6.356771 481018 .02 256.0 -6.356771 481018 .02 256.0 -6.384262 479498 -6.32 258.0 -7.219650 435529 -9.72 259.0 -8.433596 .578722 -12.75 260.0 -10.281236 306153 +16.61 261.0 +13.151341 .220005 -22.72 262.0 -17.701294 .150257 -37.34 263.0 -23.116000 .069855 -89.93 264.0 -1d.533966 .112386 -150.33 265.0 -13.737760 .205642 -166.93 267.0 -8.844408 .361227 -176.13 269.0 -10.762233 .2 | | | | |
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| 252.0 | | | - 249713 | 44.39 |
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| 266.0 -10.762233 2896cC -173.11 267.0 -8.844408 361227 -176.13 268.0 -7.617928 416016 -177.70 269.0 -6.941569 449659 -178.54 270.0 -6.722932 461162 -178.47 271.0 -6.926955 450767 -178.49 272.0 -7.573962 416121 -177.58.49 273.0 -8.770086 364331 -175.93 274.0 -10.641960 293659 -172.83 275.0 -13.52262 210565 -166.52 276.0 -18.126440 124073 -156.40 277.0 -22.730432 073026 -94.13 278.0 -17.851872 124058 -35.97.97 279.0 -13.293006 216446 -24.10 280.0 -10.349297 302368 -17.52 281.0 -8.56671 275117 -13.43 282.0 -7.283092 432360 -6.453613 473480 -6. | | | | |
| 267.0 | | | | |
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| 270.0 -6.722932 .461162 -178.47 271.0 -6.926955 .450767 -178.49 272.0 -7.573962 .416121 -177.58 273.0 -8.770086 .364331 -175.93 274.0 -10.641960 .293059 -172.83 275.0 -13.532262 .210565 -166.59 276.0 -18.126440 .124073 -156.40 277.0 -22.730432 .073026 -94.13 278.0 -17.851872 .124058 -35.97.97 279.0 -13.293006 .216446 -24.10 280.0 -10.3d9297 .302360 -17.52 281.0 -8.516671 .275117 -13.43 282.0 -7.283092 .432360 -10.25 283.0 -6.453d13 .47348b -6.44 284.0 -6.211627 .485112 -113 | | | | |
| 271.0 | | | | |
| 273.0 | | | 450767 | -178-49 |
| 274.0 | 272.0 | -7.575962 | | |
| 275.0 -13.552262 .210565 -166.59 276.0 -18.126440 .124C73 -15C.4C 277.0 -22.730432 .C7302c -94.13 278.0 -17.851872 .124058 -39.97 279.0 -13.29306 .216446 -24.10 280.0 -10.3d9297 .302368 -17.52 281.0 -8.516671 .275117 -13.43 282.0 -7.283092 .432360 -1C.25 283.0 -6.453613 .475488 -6.44 | | | | |
| 276.0 -18.126440 -124073 -150.40 277.0 -22.730432 -073020 -94.13 278.0 -17.851872 -12.058 -35.97 279.0 -13.29306 -216446 -24.10 280.0 -10.3d9297 -302360 -17.52 281.0 -0.516671 -27.517 -13.43 282.0 -7.283092 -432360 -10.25 283.0 -6.453613 -47.3480 -6.44 | | | | |
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| 278.0 -17.851872 .12d058 -39.97 279.0 -13.293006 .216446 -24.10 280.0 -10.3d9297 .302368 -17.52 281.0 -0.516671 .275117 -13.43 282.0 -7.283092 .452360 -10.25 283.0 -6.453613 .475486 -6.44 284.0 -6.211827 .489112 -1.13 | | | | |
| 279.0 -13.293006 .216446 -24.10 280.0 -10.369297 .302560 -17.52 281.0 -0.516671 .275117 -13.43 282.0 -7.283092 .452360 -1C.25 283.0 -6.453613 .475480 -6.46 284.0 -6.211627 .489112 -1.13 | | | | |
| 280.0 -10.3d9297 | | | | |
| 281.0 -8.516671 .275117 -13.43 282.0 -7.283092 .432360 -10.25 283.0 -6.493813 .473486 -6.84 284.0 -6.211827 .489112 -1.13 | | | | |
| 282.0 -7.283092 .432360 -10.25 283.0 -6.493813 .473480 -6.44 284.0 -6.211827 .489112 -1.13 | | | | |
| 283.0 -6.493d13 .4734db -6.d4 284.0 -6.211627 .489112 -1.13 | | | | -1C.25 |
| | | | | |
| 285.0 -6.341411 .48186929 | | | | |
| | 285.0 | -6.341411 | .481869 | 29 |

| 286.0 | -6.702080 | .462270 | 1.53 |
|----------------|--|--------------------------|--------------------|
| 267.G | -7.316034 | .430723 | 3.99 |
| 288.0 | -8.768273 | ,3644C7 | 14.84 |
| 289.0 | -10.388354 | • 3624CC | 26.48 |
| 290.0 | -11.904481 | .25396£ | 43.62 |
| 291.0 | -12.702476 | . 229549 | 68.70 |
| 292.0 | -12.114734 | .247852 | 54.43 |
| 293.0 | -10.596097 | .295254 | 113.71 127.15 |
| 294.0 | -8.901703 -7.526790 | .358852 .420398 | 135.91 |
| 295.0 - | -5.389804 | .479152 | 142.98 |
| 297.0 | -5.620965 | .523542 | 148.62 |
| 298.0 | -5.189398 | .55G23I | 153.55 |
| 299.0 | - ⊹ •997≥92 | .562510 | 150.40 |
| 300.0 | -5.103191 | .5557CU | 163.62 |
| 301.0 | -5.415234 | .536091 | 169.30 |
| 302.0 | -5.955818 | .503153 .466912 | 176.41 -175.32 |
| 303.0 304.0 | -6.615291 -7.398874 | .426535 | -164.81 |
| 305.0 | -8.044581 | €4004£. | -151.61 |
| 366.0 | -d. 427761 | .37897¢ | -136.44 |
| 307.0 | -8.298932 | <u>. 364639</u> | -120.51 |
| 303.0 | -7.751326 | -4C967C | -106.15 |
| 309.0 | -6.969437 | · 44 £258 | -94.38 |
| 310.0 | -6.179856 | .49091a | -84.66 -76.73 |
| 311.0 | -5.434936 | .524876 .570144 | -7C.37 |
| 312.0 | -4 . 880308 -4 . 464334 | .596113 | -64.61 |
| 313.0 314.0 | -4.217007 | . €153ES | -59.45 |
| 315.0 | -4.096583 | .623980 | -54.88 |
| 316.0 | -4.162849 | .619238 | - ∃0.02 |
| 317.0 | -4.368648 | •604739 | -45.31 |
| 31 8. 0 | -4.728238 | .580214 | -39.98 |
| 319.0 | -5. 156591 | <u>552254</u> | -33.81 -26.95 |
| 320.0 | -5.689085 -6.195409 | .519453 .49003a | -19.57 |
| 321.0 322.0 | -6.605859 | .467423 | -10.84 |
| 323.0 | -6.935850 | 445953 | -, 98 |
| 324.0 | -7.054555 | .443067 | 8.96 |
| 325.0 | -7.044222 | . 4444 15 | 18.41 |
| 326.D | -6.812815 | .456414 | 27.14 |
| 327.C | -6.508363 | •472696 •492534 | 35.16 42.01 |
| 32d+0 | -6.151280 -5.790531 | .513421 | 47.68 |
| 329.0 330.0 | -5.474178 | .5324€5 | 52.69 |
| 351.0 | -5.211626 | •5488Co | 56.67 |
| 332.0 | -5.062304 | •55 a 3 2 2 | 20.27 |
| 333.0 | -4.96168 1 | .56482d | 63.82 |
| 334.0 | -4.957960 | .565070 Earlan | 67.00 70.37 |
| 335.0 | -5.022572 -5.00006 | .560882 .549484 | 73.28 |
| 336.0 337.0 | -5.200904 -5.492511 | \$51542 | 76.47 |
| 338.0 | -5.880886 | .5081Cá | 75.95 |
| 339.0 | -6.222137 | .48853∠ | 84.15 |
| 340.0 | -6.704604 | .462136 | 88.19 |
| 341.0 | -7.324432 | .4303C4 | 92.79 |
| 342.0 | -7.959947 | - 399947 | 58-19 |
| 343.0 | -8.606558 | .371255 .344671 | 103.65 110.00 |
| 0 4.4د | -9.251912 -6.070017 | .149671 .16953 | 116.75 |
| 345.0 346.0 | -9,979017 -10,619492 | . 294459 | 123.52 |
| 347.0 | -11.311465 | .271911 | 130-27 |
| 343.0 | -11.758194 | • 25 d2 e0 | 137.64 |
| 349.0 | -12-358474 | .240756 | 142.77 |
| 350.0 | -12.891213 | 226654 | 149.0B |
| 351.0 | -13.352940 | . 214958 204417 | 154.12 158.25 |
| 3: 2.0 | -13,772675 | .204817 .192570 | 160.24 |
| 353∙0 354•0 | -14.308213 -14.809170 | .181778 | 160.79 |
| 355.0 | -15.323370 | .171329 | 160.85 |
| 356.0 | -15.711017 | 1o3851 | 1 6 C • 45 |
| 357.0 | -15.854250 | .161171 | 158.23 |
| 358.0 | -15.839272 | .161449 | 156.06 153.77 |
| 359.0 | -15.902779 -10.021174 | .160273 .158103 | 153-12 |
| 360.0 | -100 OFILL | | - - |
| | | | |

| POINT | PURT | POINT | PORT | G, MILLIMHOS | 8, MILLIMHOS |
|-------|------|-------|------|--------------|--------------------|
| 5 | 1 | 5 | 1 | 0.0000000 | 0.00000000 |
| 5 | 1 | 11 | 2 | 0.000;0000 | C. CCCCCOOO |
| 5 | 1 | 67 | 3 | 0.00000000 | 0.00000000 |
| 5 | ı | 95 | 4 | 0.00000u00 | 0.30000000 |
| 11 | Ž | 5 | 1 | C. CCC00000 | 0.00000000 |
| 11 | 2 | 11 | 2 | 0.00000000 | G. 00000000 |
| 11 | Z | 87 | دَ | 0.00000000 | C+ GC000000 |
| 11 | 2 | 93 | 4 | 0.00000000 | 0.00000000 |
| £7 | 3 | 5 | 1 | 0.00000000 | 0.00000000 |
| 87 | 3 | 11 | 2 | 0.0000000 | C. CCCC0000 |
| 87 | 3 | 87 | 3 | C. 0C0C0000 | C. CC000000 |
| 87 | د | 93 | 4 | 0.00000000 | 0.00000000 |
| 93 | 4 | 5 | 1 | C.CCG0000 | c. ccocoood |

SHORT-CIRCUIT AUMITTANCE MATRIX

On output, the input data, except for the 55 cards giving the coarse point X- and Y-coordinates, are printed first. Then, the indices and the X- and Y-coordinates of the 166 actual points generated by subroutines SPLFIT and SPFIT2 and required by subroutine TESLOT are printed in units of centimeters. The statement KNPLT = 1 implies that subroutine SKETCH has made a plot of these points and has given the scale factor below the plot in cm/inch (on the plotting paper).

C. CC000000

G_0000000 G_00000000

Next, computations and plots as functions of frequency are made. The printer output is given only for the first frequency, 5.250 GHz. The printer output however is similar for all the remaining frequencies.

For a given frequency, the frequency and wavelength are written first, and a plot of the 166 points (KNTPLT = 2) is made and the scale factor in wavelengths/inch on the plotting paper) is given below the plot. Next, a write out of the geometry of the cross section is given. In the write out, the X- and Y-coordinates are given in wavelengths at the specific frequency for a particular point, say, point I, and the length, in wavelengths, of the segment the end points of which are point I and point I + I is also given. The voltage strength of a narrow axial slot is also given at the point at which the slot is located.

Next, the computed radiation pattern is printed. In the program, PHI corresponds to $\phi_{\mathbf{r}}$ of figures 1 and 8, and for a given value of PHI, the real part, the imaginary part, the magnitude, and the phase of the relative radiation field are printed.

Next, for a given value of PHI, the radiation field in decibels, the normalized radiation field, and the radiation field are printed. The statements KNTPLT = 3 and KNTPLT = 4 indicate that subroutine DBPLOT has made a radiation field magnitude polar plot and a radiation field phase rectangular plot, both plots of which are functions of $\phi_{\mathbf{r}}$. (See fig. 14.)

Finally, the short-circuit admittance matrix is printed. However, since MADM = 0, this part of the write out is meaningless.

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TABLE I.- GEOMETRY AND EXCITATION DATA FOR APPLICATION OF RIEF FOR COMPUTATIONS OF SPACE SHUTTLE ANNULAR SLOT RADIATION PATTERNS

[All linear dimensions are for 1/35-scale model]

| Case | Figure | Perimeter, cm | Number of points and simultaneous equations | Coordinates x,y, cm, of center of annular slot | Phasor voltage strength, volts, and coordinates x,y, cm, of equivalent narrow axial slots |
|------|---------------|---|--|---|--|
| 1 | 6 | 70. 55 | 154 | $A = (0., 14.690) \text{ at } P_{43}$ | A $\begin{cases} V_{40} = 1/0^{0} & \text{at } P_{40} = (0.762, 14.690) \\ V_{46} = 1/180^{0} & \text{at } P_{46} = (-0.762, 14.690) \end{cases}$ |
| 2 | 7 | 7 0. 58 | 159 | B ₁ = (9.165, 1.600) at P ₈ | $B_{1} \begin{cases} V_{5} = 1 / 0^{0} & \text{at } P_{5} = (9.369, 0.866) \\ V_{11} = 1 / 180^{0} & \text{at } P_{11} = (8.961, 2.334) \end{cases}$ |
| 3 | 3 8 70.58 166 | B ₁ = (9.165, 1.600) at P ₈ | $B_{1} \begin{cases} V_{5} = 1 / 0^{0} & \text{at } P_{5} = (9.369, 0.866) \\ V_{11} = 1 / 180^{0} & \text{at } P_{11} = (8.961, 2.334) \end{cases}$ | | |
| 3 0 | 0 | 70. 98 | 100 | $B_2 = (-9.165, 1.600) \text{ at } P_{90}$ | $B_{2} \begin{cases} V_{87} = 1 / 180^{0} & \text{at } P_{87} = (-8.961, 2.334) \\ V_{93} = 1 / 0^{0} & \text{at } P_{93} = (-9.369, 0.866) \end{cases}$ |
| 4 | 11 | 142. 86 | 178 | $A = (0., 11.500) \text{ at } P_{48}$ | A $\begin{cases} V_{45} = 1 / 180^{0} & \text{at } P_{45} = (0.741, 11.766) \\ V_{51} = 1 / 0^{0} & \text{at } P_{51} = (-0.741, 11.234) \end{cases}$ |

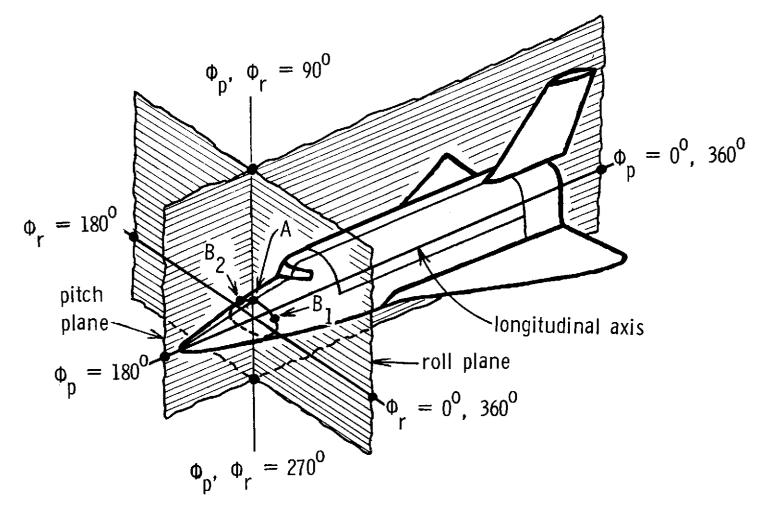


Figure 1.- Geometry of the Space Shuttle orbiter and location of the pitch and roll planes.

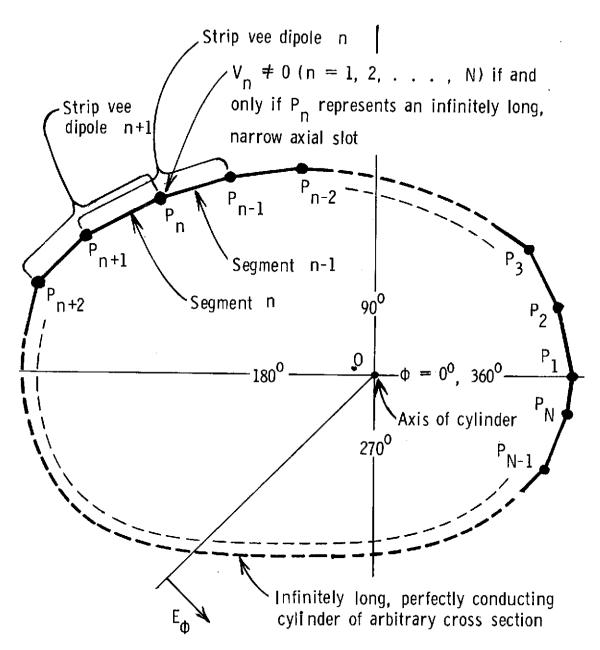


Figure 2.- Geometry for which RIEF is applicable.

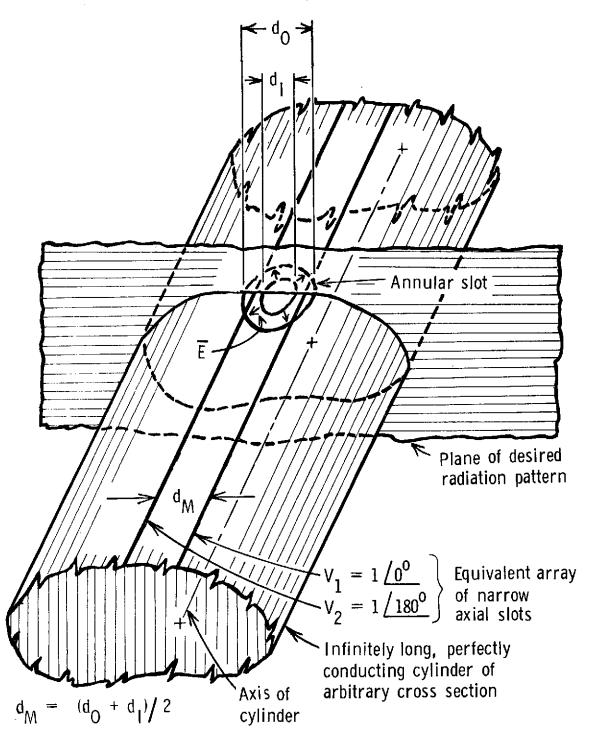


Figure 3.- Representation of an annular slot by an equivalent array of two narrow axial slots for purposes of computing the specified radiation pattern.

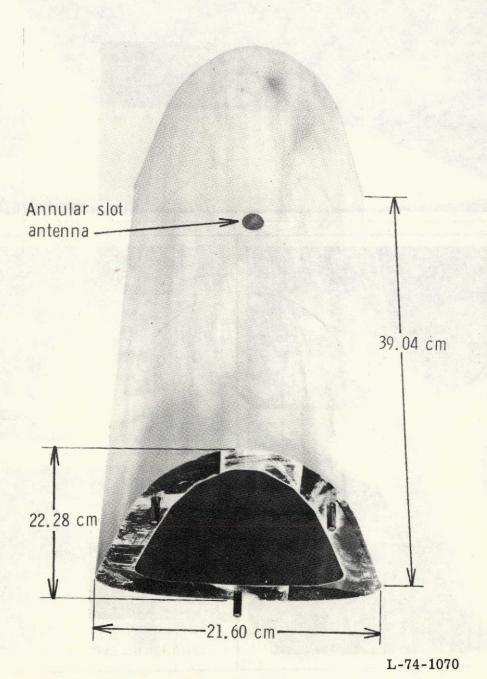
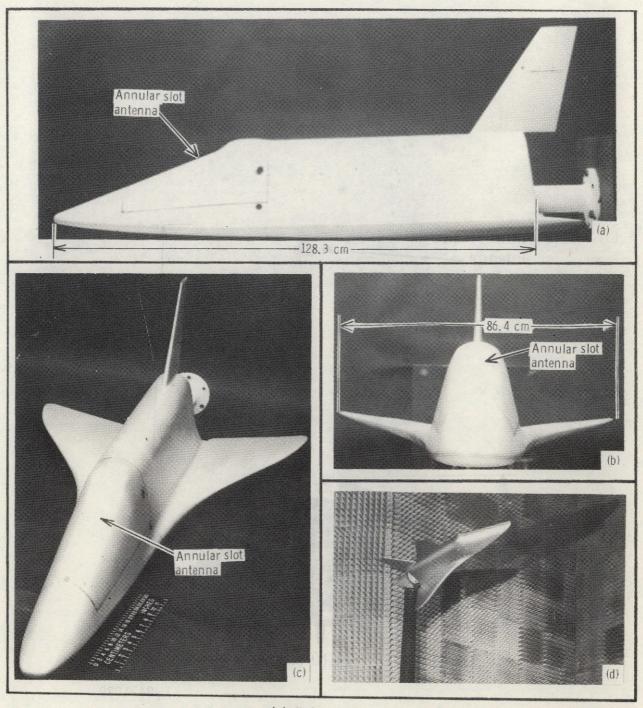


Figure 4.- 1/35-scale model (cylindrical model) for obtaining roll-plane radiation patterns experimentally.



(a) Side view.

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- (b) Front view.
 - (c) Oblique view.
 - (d) Model under test.

Figure 5.- 1/35-scale model (three-dimensional model) for obtaining roll- and pitch-plane radiation patterns experimentally.

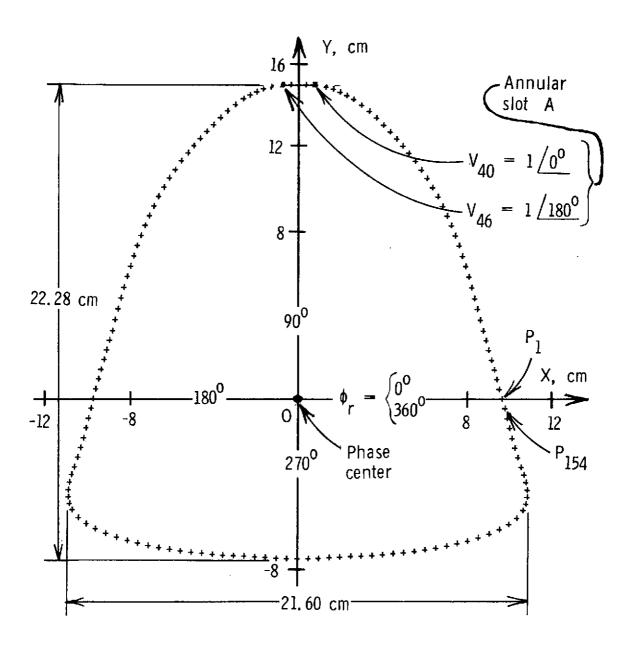


Figure 6.- Points describing roll-plane cross-section profile for case 1.

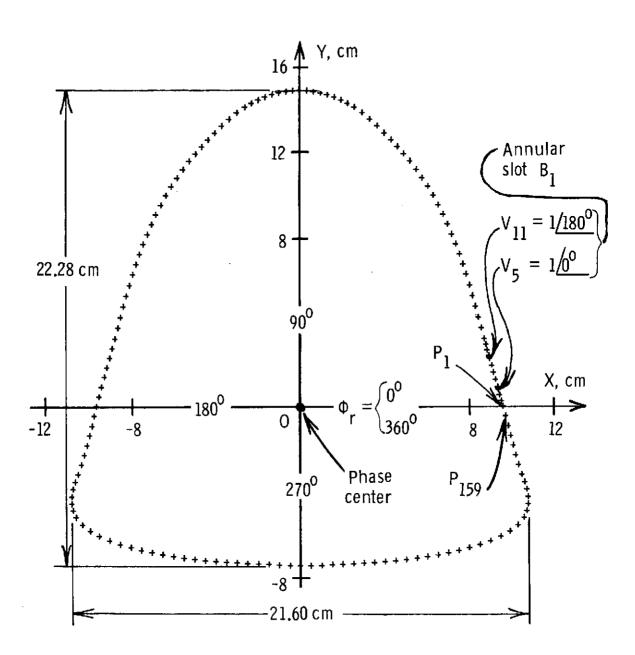


Figure 7.- Points describing roll-plane cross-section profile for case 2.

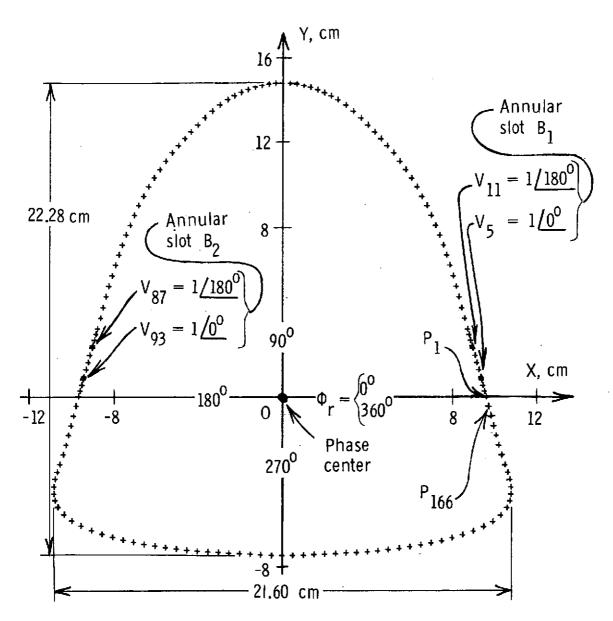


Figure 8.- Points describing roll-plane cross-section profile for case 3.

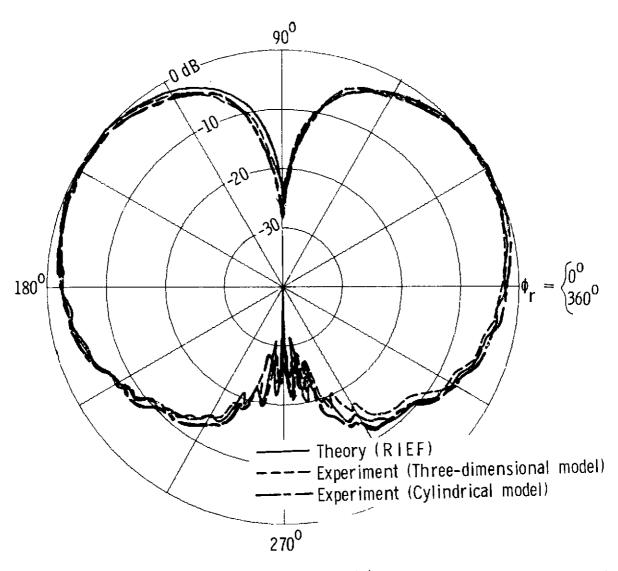


Figure 9.- Roll-plane radiation pattern for case 1 (theory compared with experiment). Frequency, 311.4 MHz (10.900 GHz).

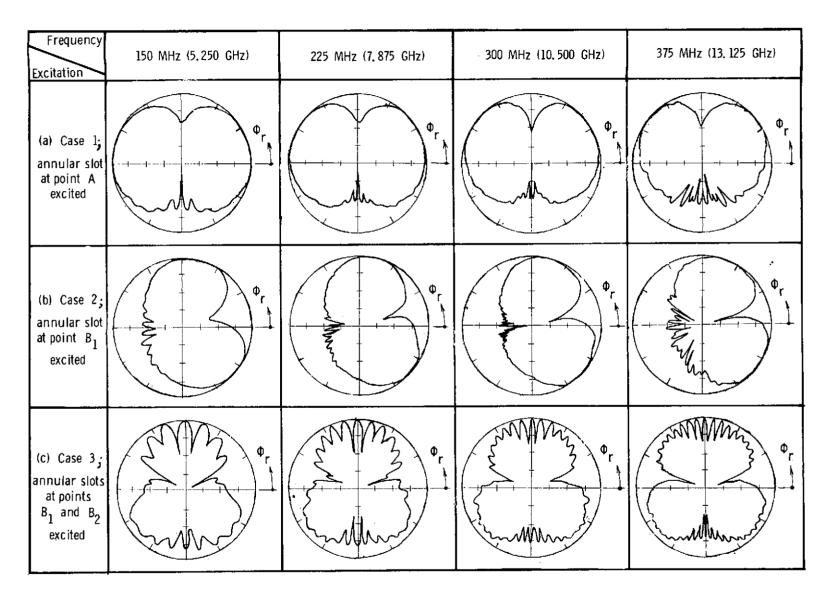


Figure 10.- Computed roll-plane radiation patterns as function of frequency.

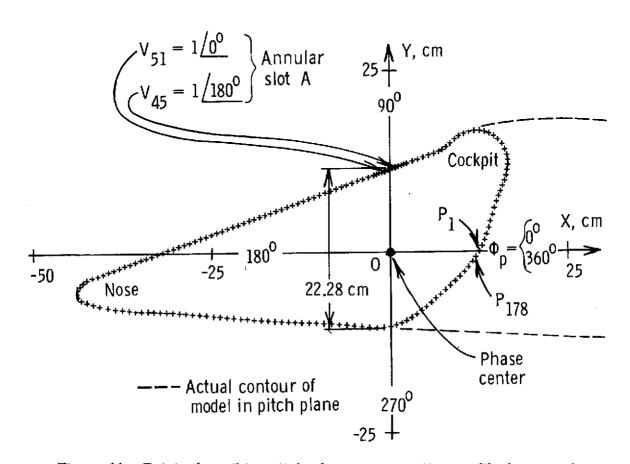


Figure 11.- Points describing pitch-plane cross-section profile for case 4.

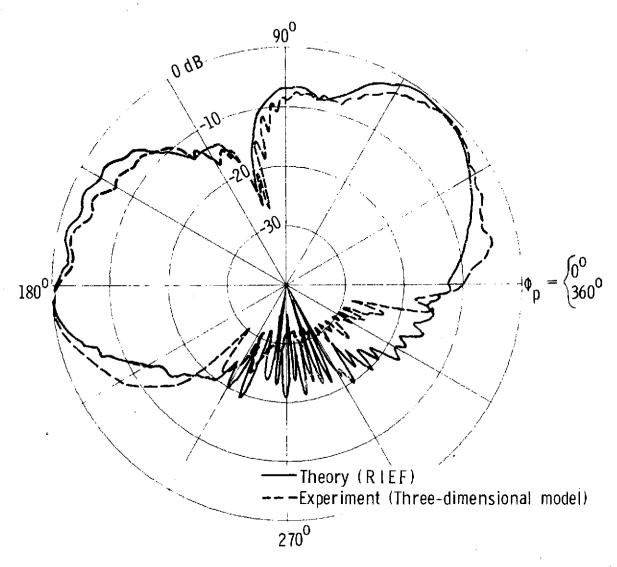


Figure 12.- Pitch-plane radiation pattern for case 4 (theory compared with experiment).

Frequency, 155.7 MHz (5.450 GHz).

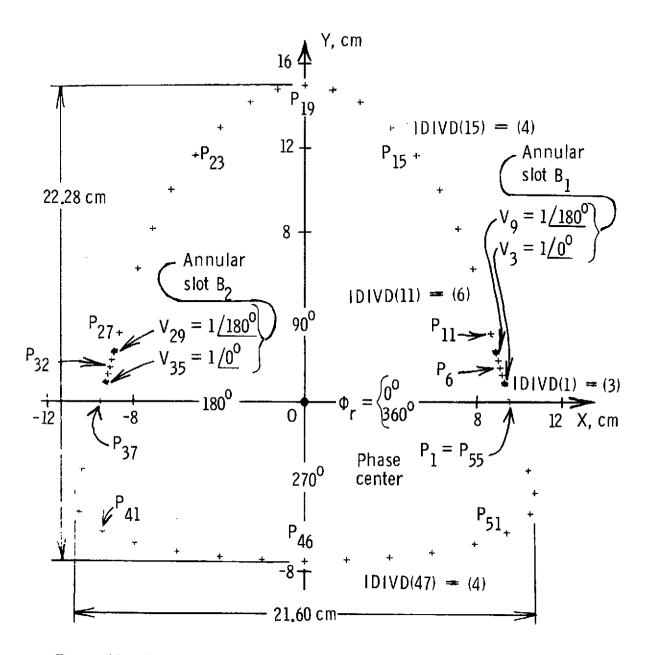
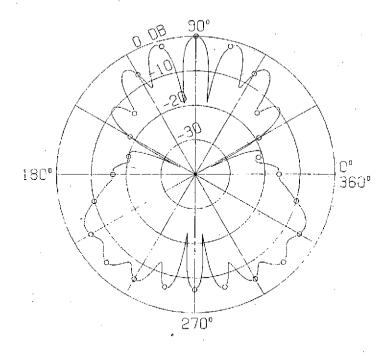


Figure 13.- A plot of the coarse points for case 3; that is, for the example given in appendix B. The coarse points are indexed sequentially as one moves counterclockwise from the positive X-axis. The IDIVD array values are given for selected cases.



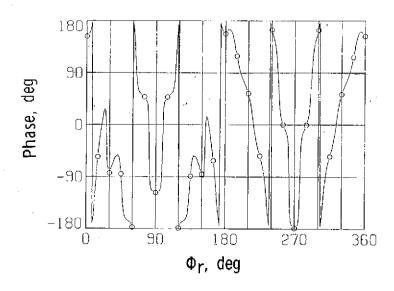


Figure 14.- Graphic output, at a frequency of 5.250 GHz, for the example given in appendix B. The polar plot is a plot of radiation field magnitude in decibels against the rollplane reference angle in degrees.